FATIGUE LIFE ASSESSMENT OF A CRITICAL FIELD WELDED

CONNECTION ON A TEMPORARY RAILWAY

BRIDGE SPAN

by

CURTIS HEINSEN

Presented to the Faculty of the Graduate School of

The University of Texas at Arlington in Partial Fulfillment

Of the Requirements for the Degree of

MASTER OF SCIENCE IN CIVIL ENGINEERING

THE UNIVERSITY OF TEXAS AT ARLINGTON

MAY 2015

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Acknowledgments

This thesis is the product of a convergence in my professional and academic careers. It has been a challenging lesson in the art of academic research and time management. Through the course of this research I have discovered and studied a number of concepts and principles that I am confident will serve me well throughout my career. However, without the support and guidance of a number of certain individuals I can say with absolute confidence that I could never have made it this far in my life either professionally or academically and I would like to dedicate this research to those people.

First among those that I must sincerely thank is Dr. Ali Abolmaali. In my time studying the science of engineering I have found Dr. Abolmaali's lessons to be the most challenging and the most thought provoking. It was because of his class in the fundamentals of the finite element method that I first began to take a real interest in the use of FEM. He has been more than generous in terms of allowing me all the resources I needed to conduct meaningful research and incredibly flexible by allowing me to conduct my research while pursuing my career. His guidance has proved invaluable and I know I would not be where I am now without his help.

I would also like to express my deepest gratitude to Dr. Mohammad Razavi who supervised my thesis and provided me with unparalleled support and encouragement. Dr. Razavi provided valuable assistance and advice from the time I began my research and his comments and suggestions have been essential to my success. I would also like to thank Dr. Leininger and Dr. Mattingly who presided over some of my first classes in engineering and who both were very supportive of my goal to attain a Master of Science degree.

I would also like to acknowledge the role of my colleagues in the field of railway engineering who provided me with their insight into my research and who have always shown a great deal of faith in my abilities. Without their valuable input this research would have proven to be much more challenging.

Lastly I would like to thank my friends and family including my two sisters and my friends Ryan and Eric who have always been around to help me in all things. Finally I want to thank my parents. I cannot say enough about how much I owe them both for all of my accomplishments and so I can only say thank you and hope that they both know how much their support has meant, and continues to mean.

Thank you all.

April 4, 2015

Abstract

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The University of Texas at Arlington

Supervising Professor: Mohammad Razavi

There is a need in the railway industry for a flexible solution to the problem of service interruptions due to impacts on railway bridges from highway and other vehicle traffic. One such solution is the placement of a temporary span that can be used on multiple substructure types and configurations and can be quickly assembled from a kit of available parts. In order to minimize the erection time and maximize the span flexibility such a span could utilize welded connections in place of bolted connections throughout the structure. These connections would be vulnerable to fatigue failure which has been shown in the AREMA design manual as well as various literature to be the primary limiting factor on the effective lifespan of most steel railway structures including the temporary span. This thesis analyzes such a structure using the finite element method

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and attempts to determine the controlling fatigue life for any number of reasonable configurations in which the span is likely to be utilized. Then a determination is made as to whether the design is valid for use in an emergency situation where a replacement structure could take months or even years to design, fabricate and construct. Phase 1 of this study leverages the finite element analysis capability of RISA 3D to determine the location of the most fatigue vulnerable connection in all of the considered configurations of the temporary span. Multiple load cases are considered including the controlling "315K" railcar. From this analysis an acceptable approximation of the real life forces the connection is subjected to can be obtained as well as a good idea of the applied forces and cycle frequency experienced by the connection. Phase 2 of this study utilizes the finite element method (FEM) functionality of ABAQUS to obtain parameters necessary for the completion of a comprehensive fatigue analysis. Parameters that were obtained during the analysis include the effective stress range (ΔS_{re}) at the toe of the fillet weld in the critical connection, the stress intensity factors for a crack at the toe of the weld (K_I, K_{II} , K_{III}), and the residual stresses induced by an arc welding procedure (σ_{res}). Phase 3 of this study utilizes the principles of linear elastic fracture mechanics to propose a procedure for the estimation of the fatigue life of the critical connection in terms of number of cycles to failure both considering and neglecting residual stress effects. Finally, the significance of this research relative to the usefulness of the temporary span is discussed and recommendations regarding how to leverage the results of this research for real world use cases of the temporary span are put forward.

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Chapter 1

Introduction

1.1 Background

Bridge strikes that cause major damage to railway infrastructure are a serious problem that the industry has had to contend with ever since the advent of the interstate highway system as a means of mass transit. Major interstate highways often follow the carefully surveyed routes established by their older railway counterparts in an effort to make use of the advantageous terrain. As such, frequent overlaps in infrastructure are inevitable. Thousands of railway bridges are at risk of being struck by semi-trucks and other oversized vehicles that fail to take proper care to check clearances before choosing their routes. Railway bridges are also vulnerable to impacts on local roads where there might be shorter spans and less stringent vertical clearance requirements. Figure 1 illustrates the damage that can be caused to railway infrastructure by such impacts. The incident in question almost certainly caused enough damage to necessitate that the bridge be taken out of service, thus negatively affecting the railroads core business. Without a viable emergency strategy the route in question would have to remain closed for several weeks as design and fabrication work were taking place.



Figure 1 - Damage from Impact

It is evident from the example outlined in Figure 1 that the railway industry could greatly benefit from further studies into rapid bridge replacement techniques and temporary strategies that include guidance on maximum applicable use cases and total lifespan of the solution. The focus of this study is to determine the maximum effective lifespan due to fatigue (controls design lifespan for steel railway bridges [2]) of a proposed temporary span that can be used in the event of a catastrophic railway bridge impact.

1.2 Literature Review

In order to complete a comprehensive study on the lifespan of a temporary structure due to fatigue it is necessary to draw on prior research taken from a number of sources that range from case studies to dissertations to text books. The literature referenced in this study covers a wide array of topics of which some of the most crucial include; research into temporary spans and rapid bridge replacement techniques, fatigue and fracture mechanics, and finite element analysis.

1.2.1 Temporary Spans and Rapid Bridge Replacement Techniques

The rapid construction of bridge systems with an emphasis on response to unexpected events has been an active field of study that has seen a great deal of additional research and attention since 2001 due to an increased awareness of the vulnerability of U.S. transportation infrastructure to catastrophic accidents or attacks. Researchers at Texas Tech University, (Burkett and Nash et al [39]) in conjunction with the Texas Department of Transportation developed a report with the aim of outlining some of the best methods that could be used to lessen the impact on the nation's infrastructure should such an accident or attack occur. The report outlines strategies and techniques for rapid bridge construction for the following components and situations;

- Superstructure techniques
- Deck techniques
- Substructure techniques
- Member/Element repair techniques
- Floating Bridges/Supports
- Contractor/Construction techniques

For the purposes of this study the section on superstructure techniques proved especially valuable as an overview for the various temporary span systems available. Some of the systems outlined by Burkett and Nash that are applicable to railway use are the Bailey bridge and the Acrow Panel Bridge system.

The work of Farah, Hunt and Pajk [39] provide an in depth case study on a project that utilized an Acrow 700XS design as a temporary span on a major railway line. This case study demonstrated tangible proof that temporary spans utilizing Acrow Panel technology could be effectively utilized to sustain railway design loads for a period of at least 5 months (time temporary span was in service). The project outlined in this case study was the replacement of a stone cut arch bridge built to carry a single track rail line in 1902. The maximum span of the arch structure was 20 feet with a maximum clearance of 12'-2" making it extremely vulnerable to bridge impacts due to vehicle traffic. It was determined to use a temporary "jump span" to allow removal of the existing bridge while still servicing rail traffic. The use of the Acrow bridge system as a temporary span cut down significantly on the amount of time the track had to be out of service as the existing bridge removal and prep work for the permanent structure took several months. Once the site was prepped for the construction of the permanent structure the Acrow 700XS was removed and disassembled in order to be used for other projects. This study also shows that spans of up to 125' designed for E-80 loads are possible with this system. Figure 2 shows the existing, temporary and permanent spans respectively. Note the extensive shoring work underway below the temporary span in preparation for the permanent structure.



Figure 2 - Existing, Temporary and Permanent Spans for Alkire Road Bridge Project [39]

1.2.2 Fatigue and Fracture Mechanics

The fundamentals of fatigue and fracture mechanics are covered extensively by Barsom and Rolfe [19] in the publication Fatigue and Fracture Control in Structures which provides the basis of many of the assumptions proposed in this study. Their work approaches the topic of fracture mechanics from an application point of view related to the field of fracture and fatigue control in structures. Chapters 1 through 6 outline the fundamental basis of linear elastic fracture mechanics including; the theoretical development of stress intensity factors (SIF), the test methods for obtaining critical stress intensity factors for intermediate loading rates, the effects of temperature, load rate and plate thickness on fracture toughness for structural materials, correlations between test methods and the Charpy V-Notch (CVN) test and the relationship between stress, flaw size and material toughness. Another publication also by Barsom and Rolfe [34] expands on the topic of test methods for obtaining critical stress intensity factors and proposes methods for estimating the stress intensity factor using CVN results. Chapters 7 through 13 ([19]) deal with sub-critical crack initiation and growth due to fatigue and corrosion. Chapter 9 presents perhaps the most relevant topic in regard to this study which is the

method for applying fracture mechanics to study fatigue crack propagation under constant amplitude cyclic loading. An in depth explanation of how the methods proposed in this publication were used in this study can be found in Chapter 6.1. Since Barsom and Rolfe focus primarily on Mode I crack growth interaction (See Figure 61) it was necessary to examine the work of Tenaka [24] in order to study the effects of multiple mode interaction (Equation 29).

The work of Martinez [13] represents an extensive study of fatigue behavior in high strength steel weldments. The study was conducted over a period of 4 years and covered topics related to steel weldments such as weld processes and imperfections, residual stresses and relaxation, spectrum loading of improved weldments and fatigue crack growth modelling. Of particular importance in relation to this study was the systematic study of imperfections associated with welding processes. Martinez presents a comprehensive weld defect comparison between various weld methods and filler metals and includes a detailed study on the average and maximum sizes for initial weld defects associated with each process. This information was leveraged in this study to propose a conservative value for an initial weld defect as a starting point to begin a linear elastic fracture mechanics analysis.

The combined works of Fett [29] [30] [31] focus on the determination of stress intensity factors using weight functions for a multitude of special cases including residual stress fields. The failure of structural members due to fracture or fatigue cracking and predicted using LEFM is governed by the stress in the vicinity of the crack tip and is

characterized by the stress intensity factor (SIF). The SIF is dependent on a number of variables including the geometry of the component as well as the loading conditions. The development of weight functions for specific problem types that are dependent only on crack geometry makes the determination of many fracture mechanics problems much simpler. The fundamentals of the weight function method proposed in these studies were combined with the information available from Barsom and Rolfe [19] in order to form the basis of understanding on the weight function method for this study.

<u>1.2.3 Finite Element Analysis</u>

It was extremely important to utilize related finite element analysis (FEA) studies as a means of comparison to ensure that reasonable results and processes were being achieved throughout this study. Research conducted by Barsoum and Jonsson [14] into fatigue assessment of cruciform joints fabricated with different welding processes using linear elastic fracture mechanics (LEFM) served as one such important reference. In this study fatigue testing and defect assessment were performed on cruciform specimens using both robotic and manual welding with flux and metal cored filler materials in order to study the effect that the welding method had on fatigue life. The study's authors then proposed a FEA methodology for determining the fatigue life of the structure using 2D finite element models to simulate a continuous cold lap induced crack and 3D finite element models to simulate cold lap defect cracks due to spatter. This study is particularly important in that it draws a direct comparison between fatigue life measurements taken from traditional physical methods of fatigue testing and results

obtained using FEA and shows conclusively that FEA can be used to accurately predict fatigue life of a welded connection.

Another interesting research study conducted by Barsoum and Barsoum [15] sought to develop a welding simulation procedure using FEM software ANSYS to predict residual stresses in T-fillet and butt welded plates. Topics including temperature fields, heat transfer, residual stresses and weld induced deformation are discussed at length and were greatly influential in the development of the proposed approach to residual stress determination in this study. The FEA results obtained in this study were also systematically compared with experimental and numerical data to once again show that FEA results show good agreement with real world results.

Lindgren [18] gives a comprehensive overview of welding simulation using the finite element method (FEM) with some attention given to using simulation to improve the design of processes and components as opposed to a means to check existing experimental data. In part 1 of this study it is shown that increasing complexity of model simulations (finer meshes, more complex boundary conditions etc.) results in more applicable results in most cases. Parts 2 and 3 outline the role of material modelling and computational efficiency of the FEM procedure. Lindgren's study is important in showing that the use of finite element simulations as a means to predict physical behavior and drive design processes has gained much wider acceptance within the field of engineering over the last decade and supports the validity of the assertions made in this study of the fatigue life of a welded temporary span connection.

Radaj [25] provides another excellent resource for the study of welded residual stresses using FEM. This study discusses in great detail the decoupling of the temperature and stress fields during an analysis which is achieved by feeding the heat transfer values into a static stress analysis as a loading history (See section 5.2.3 for more details). Rajad includes typical examples of the calculation of the temperature stress field and residual stresses in the study which can then be utilized as a comparison to ensure that reasonable FEM results are obtained in future studies.

1.3 Overview of Alternative Efforts to Implement Rapid Bridge Replacement Techniques

In the past, a number of strategies have been studied and implemented in order to facilitate the rapid construction of replacement structures for both highway and railway traffic with varying degrees of success. This section seeks to outline some of the methods that are most applicable to the replacement of a damaged railway span due to catastrophic impact damage and assess the relative strengths and weaknesses of these approaches.

1.3.1 Span by Span Placement Method Using Existing Material

As the name suggests the Span by Span Placement Method (SSPM) is a method of constructing a bridge sequentially starting at one abutment and progressing span by span to the opposite abutment using the previously constructed span as a platform to build the next. For railway applications it is typical practice to utilize an on-track crane to set the spans. This is one of the most common methods used to construct replacement structures for aging railway infrastructure as it is a very efficient means of constructing a

bridge within a narrow window of time (sometimes as few as six hours from start to finish) [37]. This same methodology utilized for planned bridge replacements can be adapted for use in an emergency situation such as the scenario referenced in Figure 1.

It is common among many of the larger railway companies to maintain a constant stock of bridge material. As ballast deck pre-stressed concrete spans have become the most common construction type for new railway bridges less than 50 feet, this stock is usually comprised mostly of pre-stressed concrete box girders and slabs with spans ranging from 14 to 45 feet and stored in fabrication facilities located across the United States and Canada [35]. In the case of a bridge impact a suitable length span, or combination of spans, can simply be requisitioned from this stock material or purchased from another railroad company and constructed using the SSPM. In cases where a combination of spans will be needed to match the span length of the damaged bridge component a temporary bent can sometimes be erected to accommodate the new geometry. Figure 3 demonstrates the application of the SSPM for an emergency situation. Assuming that a suitable span replacement were to be located, the amount of time that an impacted bridge would be out of service could be as short as the time it takes to ship the bridge material plus around half a day to set the span.



Figure 3 - Span by Span Placement Method for an Emergency Application

Advantages of the SSPM for emergency bridge replacement;

- Utilizes available material to quickly restore damaged bridge to service.
- Leverages familiar techniques and practices to construct the replacement span.
- Since the pre-stressed concrete material is designed and fabricated to standard specifications the opportunity exists to leave the new spans as the permanent replacement structure.
- No lateral or longitudinal forces or unbalanced bending moments are introduced into the piers as would be the case with an incremental launching or balanced cantilever construction method [36] [40].

Disadvantages of the SSPM for emergency bridge replacement;

• Only tangent substructure can be accounted for using commonly available material.

- Most on-track cranes utilized by railroads are only capable of setting standard prestressed concrete spans up to 34'-0" [37].
- Geometry of roadway underpasses or other existing infrastructure may not allow for the placement of temporary or permanent bents.
- Exact dimensions of the damaged railway span may prove to be impossible to match using stock material.
- In many cases custom bearing elements such as riser blocks would need to be designed and fabricated before setting the new spans.

1.3.2 Prefabricated Modular Temporary Span Method (Acrow Panel Bridge)

Another viable solution that has been used successfully to rapidly restore and maintain train traffic during a bridge outage is the use of a prefabricated modular temporary span. While the rapid replacement of damaged superstructure elements due to impact has not been the primary focus of research and development of this strategy the core technology involved is directly applicable to this use case [38]. Of all the possible systems that are available, perhaps the most applicable to the needs of the modern railway are the Acrow Panel Bridges. Figure 4 shows the plan and elevation of an Acrow 700XS modular bridge for a temporary railway application. These versatile bridges are a modern improvement of the old Bailey bridge design developed during the Second World War to quickly erect structures to move troops and equipment [38].

An excellent example of how an Acrow Panel Bridge can be used in a temporary capacity to support railway traffic is the Alkire Road widening and railway bridge replacement project undertaken in Franklin County Ohio during May of 2011 [39]. Due to the need to limit interruptions to rail traffic at this location it was decided that the best possible course would be to construct a temporary 125' Acrow 700XS modular bridge lifted into place over the existing structure in order to allow work to be performed underneath. The modular bridge was constructed on site and took approximately 10 days to assemble utilizing a crew of just six men (See Figure 5). Setting the span was accomplished using an off-track crane and took approximately 1 hour (See Figure 6). Including the time needed to re-lay the track over the temporary span the rail line in question was returned to service in under 10 hours [39]. Due to the modular nature of construction it is reasonable to assume that with a larger crew of workers the assembly phase of construction could be achieved even faster in an emergency situation.



Figure 4 - Plan and Elevation of Acrow 700XS Customized for Railway Loads [39]





Figure 5 - Acrow Bridge Assembly
Figure 6 - Acrow Bridge Crane Pick
While the project featured in Figure 5 and Figure 6 leveraged an off-track crane to
set the temporary span, one of the primary advantages of the Acrow bridge system is that
it is designed to withstand loads from a number of different span placement methods
[40]. Some of the methods that have been used in the past include the Incremental
Launch Method (ILM) and the Barge Construction Method (BCM). ILM would be of
particular interest for an emergency replacement of a span greater than 50' and is a
method by which a span is either set on rollers or a hydraulic jacking system and pushed
or moved into place. Figure 7 demonstrates how an Acrow Panel Bridge can be set using
the ILM for a railway application [40].



Figure 7 - Incremental Launching Method Using Acrow 700XS [40]

Advantages of Prefabricated Modular Temporary Bridge Design;

- Relatively long spans of up to 125' can be achieved with this system [39].
- The ability to disassemble the span and reuse the material for other projects [40].
- The ability to add and subtract panels allows for flexibility in span lengths.
- Design allows for a greater clearance for roadway underpasses than could be achieved with beam spans or girders in most cases.
- Maximum span length allows for possibility to use as a jump span while commencing prep work for a permanent structure.
- Stock material available for purchase or rent on short notice.
- Ability to use ILM for longer spans makes on-track construction easier where access proves to be an issue.

Disadvantages of Prefabricated Modular Temporary Bridge Design;

- Clearance between modular panels can be limiting for certain types of rail traffic.
- Not all span lengths can be achieved using stock material.
- Assembly times of as much as 10 days [39] could prove to be costly in emergency situations.
- Skewed substructure would require a detailed design for each case [40].

It is clear that the alternatives outlined in this chapter represent viable solutions to the problem of service outages due to severe impacts to railway bridge infrastructure. However an examination of the disadvantages posed by the use of each system shows

that additional approaches could prove valuable to service the use cases that the proposed alternatives cannot achieve. It is obvious that the SSPM utilizing available material would be ideal for situations where available material matched well with existing geometry in a way that would allow the permanent replacement of the bridge in a short period of time. The modular temporary span method would work well in situations where the engineer is faced with damage to a long span over a major road or waterway where placing temporary bents is not an option. In between these scenarios there is a need for a span of up to 55' (a large percentage of low clearance bridges fall into this range) that can accommodate skewed substructure geometry and be erected in the field in a minimum amount of time. Utilizing an emergency steel span made from a kit of parts held in strategic locations throughout the railway system and constructed using field welds could achieve this goal and would mean that in situations like the one shown in Figure 1, train traffic could be restored very quickly. Instead of days or weeks that it might take to restore the route using alternative methods it would be possible in most cases to restore service in less than 1 day.

Chapter 2

Design of Temporary Structure

The temporary structure was designed with the express purpose of allowing the railroad a quick, safe and flexible solution to the problem of span failures due to vehicle impacts. The expected short duration of the span's service allowed for the use of field welded connections within the tension zone of diaphragms and stiffener plates that would be unusual in a permanent structure. Serviceability requirements regarding the maximum deflection were reduced from the recommended L/640 in the AREMA design manual and taken per internal emergency response guidelines as;

$$\delta_{max} = \frac{L}{460} \tag{Equation 1}$$

In all other ways the temporary span was designed in accordance with the codes outlined in the 2013 edition of the AREMA Manual for Railway Engineering, Volume 2. Some of the parameters used to define the scope of the design are listed as follows;

- Simple Span
- W36 rolled beams (based on average necessary clearance requirements)
- Steel yield strength of $F_y=50$ ksi (ASTM A572 Gr. 50)
- 4 beams per track

- Designed to withstand Cooper's E-65 loading
- Full diesel impact considered
- Span capable of supporting timber ballast deck with 8" of ballast
- $\underline{\sigma}_{max} = .55 F_y$
- Maximum degree of curvature to be considered as 2 degrees
- Maximum allowable speed of rail traffic to be considered as 45 mph

The Cooper E loading that was used in the design of this temporary span, as well as most modern railway bridges, is based on two 2-8-0 Consolidation-type steam locomotives with an infinite number of railcars represented by a uniformly distributive load (see Figure 8 and Figure 9) [7] [1]. This standard was first proposed in 1894 by Theodore Cooper to the American Society of Civil Engineers as a load pattern that would accurately mimic the controlling forces exerted on bridges from railway traffic at that time. The loading system is set up as scalable with the original magnitude of the heaviest axles considered in the proposal to the ASCE by Cooper in 1894 set to E-10 and most permanent modern railway bridges designed for E-80. The Cooper's E loading seems to be outdated, especially in terms of the steam locomotive geometry, but with the ever changing and varied geometries seen in modern use the Cooper's loading has proven to be a good (conservative) approximation of the maximum stresses experienced by modern structures under load [2] (*Figure 31 and Figure 32 in the section on live loads illustrate how Cooper's E-80 load is conservative for maximum stresses in most cases).



Figure 8 - 2-8-0 Consolidation-type Locomotive - Basis for Cooper's E loading [8]



Figure 9 - Diagram of Cooper's E-80 Loading for Modern Railway Bridges [1]

Sections 2.1 through 2.7 outline the results from an extensive design and analysis performed to determine the optimum span geometry for the emergency structure. The

beams were sized for maximum Cooper E-65 forces on a 55 foot configuration. The entire design process included a detailed cost analysis, logistics study and field action plan that were decided to be outside the scope of this study. Figure 10 shows the basic plan of the temporary span.



Figure 10 - Plan of Temporary Span

The design analysis results presented in the following sections are valid for an ASTM A572 steel beam span consisting of 4~W36 X 262 rolled I beams on track curves not to exceed 2 degrees and speeds not to exceed 45 mph. Furthermore the design stipulates that the span shall not be utilized in cases where the substructure skew is

greater than 30 degrees. For details on placement of stiffeners and diaphragms please examine Figure 10.

2.1 Dead Load (E-65)

The calculation of the dead load for most railway applications can typically be estimated with a reasonable degree of accuracy due to the mostly standardized materials used. For the purposes of design for the temporary span the following items were considered; the self-weight of the beams, diaphragms, stiffener plates, hardware, rails, plates, ballast and timber. While the calculation of the dead load included the weight of the ballast and timber that would be typical in a ballast deck application it is important to note that the dynamic impact force reduction allowed by AREMA of 10% was not used due to the unknown nature of the future dead load on the span. This 10% reduction assumes a substantial dampening effect occurs between the ballast and the span and would have been inappropriate for an open deck. This assumption should prove to be conservative for the dead load capacity of the temporary span.

For ordinary steel railway bridges the dead load component is often a small percentage of the total load [2]. This is due to the fact that the live load component (locomotive/railcars) of the total load is much greater than is typical in most other structural design problems. Figure 11 and Figure 12 show the values for moment and shear due to dead load at intervals up to mid-span (*Span is symmetrical). The line labeled "Girder" in Figure 11 and Figure 12 represents the moment and shear reaction due to dead load while SDL denotes the reactions due to the sum of all the dead loads.


Figure 11 - Moment in Beams Due to Self-Weight and Superimposed Dead Load



Figure 12 - Shear in Beams Due to Self-Weight and Super Imposed Dead Load

2.2 Live Load (E-65)

The design live load of most modern railway structures is based on the Cooper E loading as previously discussed. Utilizing this design loading as a standard for the industry allows for a more simplified standard approach for design engineers while still providing a loading system that is conservative in most cases. It is of particular importance that the live load be conservative or at least accurate at predicting actual load values because the live load represents a much greater percentage of total load than any of the other load categories prescribed by AREMA. The live load considered for the temporary span was Cooper's E-65 loading whose geometry is shown in Figure 9. Figure 13 and Figure 14 show the shear and moment induced by E-65 loading on the temporary span.



Figure 13 - Moment Induced by Cooper's E-65 Load



Figure 14 - Shear Induced by Cooper's E-65 Load

2.3 Combined Factored Loads (E-65)

In addition to the dead load and live load a number of other design loads were considered during the design of the temporary span as outlined by AREMA. The fundamental basics of these additional loads will be covered more extensively in Section 3.1.3 through 3.1.5 and include impact load, longitudinal load and centrifugal load. Of all the additional loads it is impact that plays the most important part in increasing the design stresses in the structure. As discussed briefly in Section 2.1 the full diesel impact (no 10% reduction) on the span was considered in addition to the weight of the ballast deck thus ensuring a conservative result for whatever deck type is used on the temporary span. Figure 15 and Figure 16 show the moment and shear values for the temporary span due to the combined loads (including dead and live load) plus full impact.



Figure 15 - Moment Due to Total Combined Loads + Impact



Figure 16 - Shear Due to Total Combined Loads + Impact

2.4 Tension Stress Check

Figure 17 shows the calculated tensile stress in the extreme fiber of the W36x262 beam plotted against the AREMA allowable stress for steel members of [1],

$$\sigma_{all} = .55F_y \tag{Equation 2}$$

Where,

 σ_{all} = Allowable stress = 27.5 for ASTM A572 Gr. 50 steel.

 F_{y} = Material yield strength = 50 ksi for ASTM A572 Gr. 50 steel.

Examination of Figure 17 shows the temporary span is sufficient in tension per AREMA guidelines for E-65 loading.



Figure 17 - Tension Stresses vs Allowable Stress

2.5 Compression Stress Check

Figure 18 shows the calculated compressive stress in the extreme fiber of the W36x262 beam plotted against the AREMA allowable stress for steel members shown in (Equation 2). This highlights the absolute maximum stress that is allowed in the span per AREMA guidelines.

In order to account for the maximum distance of diaphragms for the purposes of preventing lateral torsional buckling the following equation was used to account for a maximum allowable compression stress (F_{call}) as a function of the unbraced length and the radius of gyration of the member's weak axis [1];

$$F_{call} = .55F_y - \frac{.55F_y^2}{6.3\pi^2 E} \left(\frac{l}{r_y}\right)^2 < .55F_y$$
(Equation 3)

Where,

l = Unbraced length

 r_{y} = minimum radius of gyration of the compression flange

Application of (Equation 3) in conjunction with special provisions in the AREMA Manual show that a maximum spacing of 11'-0" center to center is appropriate for the diaphragm system.



Figure 18 - Compressive Stresses vs Allowable Stress

2.6 Shear Stress Check

Figure 19 shows the calculated shear stress in the web of the W36x262 beam plotted against the AREMA allowable shear stress (τ_{all}) for steel members shown by the following equation [1];

$$\tau_{all} = .35F_y \tag{Equation 4}$$

This highlights the absolute maximum shear stress that is allowed in the web of the rolled beam section per AREMA guidelines.



Figure 19 - Shear Stresses vs Allowable Stresses

2.7 Initial Overview of Fatigue for Temporary Span

The above results show that the span geometry as outlined in the design is adequate to meet service requirements in the short term. Even though it is shown in later sections of this research that Cooper's loads do not necessarily accurately reflect fatigue loading in steel railway bridges it is still useful to consider fatigue effects induced by design loads as a validation for further research into the subject. The AREMA design manual offers guidance on fatigue design by classifying connection details into categories whose allowable stress ranges (ΔS_R) have been determined by empirical analysis [1]. An inspection of the stiffener plate-to-flange and stiffener plate-to-web connections indicates that these connections fall into fatigue category C¹ (see Figure 20).

¹ AREMA designates fatigue categories for common construction details and includes guidance on the maximum allowable stress range for each category. If a detail covered by a fatigue category experiences a stress range within this limit it will be able to sustain more than 2 million cycles and is considered to have infinite fatigue life [1].

In order to expedite the initial fatigue analysis only in-plane forces were considered in the rolled I-beams. Forces induced by differential deflection of the beams and transferred through the diaphragms were not considered. The effective stress range of incremental sections along the span was calculated and compared with the allowable stress ranges for the fatigue categories outlined in the AREMA manual to determine which category would control at that section (see Table 1). The manual stipulates that stress ranges that fall within the specified stress range for the appropriate category will be adequate for over 2 million cycles when subjected to continuous unit trains with axle loads not exceeding 80,000 pounds on spans less than 100 feet [1]. Table 2 shows the calculated stress range at the location of the controlling welded stiffener. Investigation of Table 2 shows that the stiffener-to-beam connection exceeds its controlling category's allowable stress range (C') by almost 2.43 ksi. In Table 2 allowable and actual stresses have been highlighted in red for clarity.

Steel Fatigue Tension Stresses on Net Section							
		Fatigue Category					
Х	$\Delta S_{r-fatigue}$	А	В	С	D	E	
(ft)	(ksi)	(ksi)	(ksi)	(ksi)	(ksi)	(ksi)	Limiting Category
0.00	0.00	24.00	16.00	10.00	7.00	4.50	Category E OK
2.70	3.11	24.00	16.00	10.00	7.00	4.50	Category E OK
5.40	5.68	24.00	16.00	10.00	7.00	4.50	Category D OK
8.10	7.84	24.00	16.00	10.00	7.00	4.50	Category C OK
10.80	9.84	24.00	16.00	10.00	7.00	4.50	Category C OK
13.50	11.45	24.00	16.00	10.00	7.00	4.50	Category B OK
16.20	12.79	24.00	16.00	10.00	7.00	4.50	Category B OK
18.90	13.63	24.00	16.00	10.00	7.00	4.50	Category B OK
21.60	14.39	24.00	16.00	10.00	7.00	4.50	Category B OK
24.30	14.68	24.00	16.00	10.00	7.00	4.50	Category B OK
27.00	14.75	24.00	16.00	10.00	7.00	4.50	Category B OK

Table 1 - Fatigue Stress Ranges for Multiple Sections of Temporary Span - Tension Zone

Steel Fatigue Tension Stresses on Net Section, At Welded Attachment							
		Fatigue Category					
Х	f _{fatigue}	А	В	C'	D	Е	
(ft)	(ksi)	(ksi)	(ksi)	(ksi)	(ksi)	(ksi)	Limiting
							Category
22.00	14.43	24.00	16.00	12.00	7.00	4.50	Category
							B OK

Table 2 - Fatigue Detail For Critical Connection 55' Span

Description	Category	Constant A (ksi ³)	Threshold S _{Rfat} (Ksi)	Potential Crack Initiation Point	Illustrative Examples
	SECTION 4	4 - WELDED	STIFFENER	CONNECTION	18
4.1 Base metal at the toe of transverse stiffener-to-flange fillet welds and transverse stiffener-to-web fillet welds. (Note: includes similar welds on bearing stiffeners and connection plates.)	C'	44 x 10 ⁸	12	Initiating from the geometrical discontinuity at the toe of the fillet weld extending into the base metal	

Figure 20 - AREMA Fatigue Category C' [1]

The results of the initial analysis indicate that the service life of the span will likely be less than 2 million cycles. Since the span is designed as a temporary measure with a limited service life this limitation may not prove to be an issue. However, now that it has been shown that the design of fatigue prone connections in the temporary span do not conform to the recommended practices accepted by the industry it becomes necessary to investigate the fatigue damage accumulated at these connections further in order to determine a safe limit to the emergency span's service life.

2.7.1 Use of Finite Element Analysis to Achieve Meaningful Results

Traditionally, fatigue analysis has been a mostly test based endeavor and therefore has been very costly as well as time consuming and labor intensive. A purely test based fatigue analysis is also problematic in that it does not allow for the flexibility to slightly change certain design parameters in order to maximize design efficiency. By utilizing a finite element analysis approach to the problem of fatigue testing for welded connections on a temporary railway span it is possible to get a similarly useful result without the drawbacks associated with traditional lab testing.

All structural analysis whether lab based or FEM requires at least three independent criteria; material, loading and geometry. Material is what the structure is made out of and it is important to understand each material's unique properties in order to understand how that material will resist loading. The materials used in the construction of the temporary railway span include ASTM A572 Structural Steel for the beam spans and diaphragms, Southern Yellow Pine for bridge ties, as well as ASTM A307 Hook bolts and hardware. The material properties being considered are independent of the method used to test them. This means that in theory as long as quality control procedures have been adequately administered on physical members their properties should be identical or within negligible tolerance compared with material properties used in an FEM analysis. The benefit of using an FEM analysis is that these properties can be changed and manipulated in order to achieve maximum efficiency.

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Loading represents the forces that the structure will be subjected to. In the case of the temporary beam span these include the structures self-weight as well as external loads such as the locomotive, rail cars, wind and seismic loads. For the benefit of a finite element analysis the loads that the structure being analyzed is subjected to can be characterized into the following cases; dead load, live load, impact load, wind load, centrifugal forces, longitudinal load and lateral forces. These forces can be carefully calibrated either through mathematical models or from empirical observation to closely mimic real world loading conditions.

Geometry represents the shape of the structure. It is intuitively understood that different shapes will be more efficient than others at resisting loads. For instance, it is understood that a piece of paper that is rolled into a cylinder will be much more effective as a column supporting a load from above that the same piece of paper stood up on its side. While the concept is easily understood, achieving the most efficient geometry in a given structure can be a difficult task. One of the major drawbacks of using physical prototypes of members to conduct lab tests (including fatigue analysis) is that in order to test a different geometry it would be necessary to construct an entirely new prototype in most cases. FEM modeling allows for a much greater flexibility in terms of geometry by allowing the user to alter or completely change geometries being tested with relatively minimum effort and without the constraining costs of building new physical models.

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Figure 21 helps to illustrate schematically how FEM modeling is leveraged in this research as a means to achieve similarly useful objectives compared with traditional labtest based analysis



Figure 21 - FEM vs Physical Modelling

When considering the problems imposed by the analysis of the temporary railway span it is easy to see how FEM modeling is by far the most attractive option. The entire premise of the structure is that it will only be used in the event of an emergency. This structure is never intended to be used unless absolutely necessary and does not utilize the traditional construction methods that have been proven safe and effective for normal railway use. The fatigue vulnerable welded connections in particular can be seen as case and point. Since there are no similar connections currently in service among class I^2

² The designation of a Class I railroad is defined by the Surface Transportation Board as a railroad having annual operating revenues of \$250 million or more after adjusting for inflation using the Railroad Freight Price Index developed by the Bureau of Labor Statistics. As of 2011 this figure stood at \$433.2 million.

railroad infrastructure that are known to this author a systematic study of the fatigue life of such a connection over time in the field is impossible. By using an FEM analysis to predict the effective fatigue life of these connections the railroad industry can be given a reasonably accurate prediction of the structures useful lifespan in a timely and cost effective way without the need to compromise safety which has always been the industries' top priority.

There are currently 7 class 1 freight railroads operating in the United States including; BNSF Railway, Union Pacific Railway, CSX Railway, Canadian National Railway, Canadian Pacific Railway, Norfolk Southern Railway and Kansas City Southern Railway. Together these Class I railroads make up the vast majority of railway track and infrastructure in the United States.

Chapter 3

Modelling of Temporary Span in RISA 3D

RISA 3D was used to model the entire temporary span in order to predict the applied forces necessary for a fatigue analysis. RISA 3D was chosen as the best candidate for the modelling of the structure for a number of important reasons. Chief among these reasons was the intuitive nature of the user interface that made modelling the temporary structure in multiple configurations relatively quick. The moving load feature available in RISA 3D was another factor that made its use an appealing option as load patterns could be set and then simply scaled up and down to simulate lighter loads (286K and 315K railcars are a good example). Some of the drawback to using this software is the inability to customize surface and part interaction or to model complex local geometries. That is why it was decided that for the purposes of this research that RISA 3D would be used to quickly test loads and span geometries to ascertain the suitability of those variables for further analysis. Once a controlling combination of loads and geometries were obtained it would then be possible to apply the forces and conditions predicted by the RISA analysis to a more detailed model of the complex geometry of the critical connection.

One of the factors often left out in the design of railway bridges is the effect that the rail and ties will have on the distribution of loads to the superstructure. It is common practice to consider the Cooper load outlined previously as acting directly on the load

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supporting element as a concentrated force. However, anyone with a background in engineering would recognize that the ties and especially the rail are both serving to distribute the load over a greater area. This is particularly true of longitudinal forces, of which a significant percentage would be transferred by the rail and not affect the substructure or superstructure of a relatively short bridge. While a complex discussion of how force distribution affects longitudinal forces is outside the scope of this research the effects of distribution of vertical and transverse forces will be captured by the finite element model as long as proper detailing of the model is achieved. In order to achieve this objective the rail was modelled as an arbitrary rectangle but assigned the section properties of 136# Class 1 rail which is one of the most common rail types on North American class 1 railway mainlines. Figure 22 shows the section properties that were assigned to the rail in the FEM model.





Area (in ²)	13.32
Section Modulus – Head	23.78
(in ³)	
Section Modulus – Base	28.3
(in ³)	
Moment of Inertia (in ⁴)	93.7

Figure 23 - 136# Rail Fastened with Pandrol

Plates

Figure 22 - 136# Rail Properties

Since all forces with the exception of wind or seismic forces act on the structure through the rail it is important that the rail-to-tie and tie-to-superstructure connection be modelled as accurate to real world conditions as possible. This created somewhat of a challenge in RISA 3D as the basic functionality is set up to connect members at nodes that are always located at the centroid of the member. In order to facilitate a "stacked" geometry where members interact from their respective surfaces it was necessary to utilize rigid links. These links were modeled with a high stiffness and zero density so as not to interfere with dead load calculations. In the case of the emergency span modeled in RISA 3D the interactions were modelled using rigid links with end conditions treated as either fully fixed for mostly rigid connections or pinned for moment released connections.

The vast majority of rail-to-tie connections on bridges are made using Pandrol tie plates that act as "elastic fasteners" and hold the rail, plate, and tie together in a mostly rigid manner. Figure 23 shows a picture of a bridge deck that utilizes Pandrol plates for rail-to-tie connections. These connections were considered capable of imparting moment to the ties and therefore a fully fixed rigid link like the example shown in Figure 24 were used to model the interactions. The tie-to-superstructure link is almost always made using either HCP clips or hook bolts like the one shown in Figure 25. It is obvious that this type of connection would be able to resist translation in the vertical (Y) and lateral (Z) direction but would provide negligible moment capacity. Therefore the tie-to-superstructure link was modelled as moment released. This has the added benefit of being conservative in terms of the stresses in the beams as it lowers the stiffness of the system.

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Figure 24 - Rigid Link for Rail-to-Tie Connection Figure 25 - Typical Hook Bolt

3.1 Forces

Once the FEM models for each span were accurately modeled in a way that most closely resembled real world connections it was important to ensure that the appropriate loads were then applied in such a way that produced the stresses and forces within the structure that correspond to the actual response. The determination of appropriate loading is a function of intense research by academics in the railway engineering field as well as a proper application of design constraints. External forces were applied at increments along the structure and solved statically with member forces and stress cycles determined from the results. In order to account for the dynamic elements of the moving train loads a series of additional loads were considered to replace such effects on the structure. This method of using static solutions to model dynamic behavior has been shown through research conducted by the American Association of Railroads to be accurate or conservative at predicting fatigue life of railway structures [1].

3.1.1 Dead Load

The dead load of the emergency span consists of the weight of the structure itself in conjunction with any permanent fixtures to the superstructure such as hardware or ties. The American Railway Engineering and Maintenance of Way association provides guidance on appropriate dead load values (See Table 3) [1]. The dead load considered for the FEM analysis is as follows.

- 4 ~ W36 x 262 ASTM A572 steel beams
- 2' x 2'-6" x ¹/₂" ASTM A572 steel diaphragm plates (varying quantities)
- 7" x 2'-10" x ¹/₂" ASTM A572 steel stiffener plans (varying quantities)
- 10" x 10" Southern yellow pine bridge ties
- 136# rail (Figure 22)

Table 3 - Dead Loads on Steel Bridges

Dead Loads on Steel Bridges				
Туре	Pounds per Cubic Foot			
Steel	490			
Concrete	150			
Sand, gravel and ballast	120			
Asphalt-mastic and bituminous macadam	150			
Granite	170			
Paving Bricks	150			
Timber	60			
Track rail, fastenings and guard rails	200 Pounds per Linear			
	Foot			

*Information from the American Railway Engineering and Maintenance of Way Association, *Manual for Railway Engineering*, Chapter 15.

3.1.2 Live Load

The determination of live loads for railway bridges can be somewhat complex due to the variability of locomotives and equipment used on railroads. The loads that existing railway infrastructure is expected to support has traditionally increased over time and the implementation of heavier loads has accelerated dramatically in the last 25 years [3]. This means that in the future the emergency beam span may be asked to support loads that it was not designed for. A railway engineer in the future that is confronted with a bridge failure and has the emergency material on hand may make a determination to allow heavier loads on the span. This is not an unreasonable decision considering that the allowable stress in the steel span is a very conservative .55f_y. However, such a decision must be made with an understanding of how the critical connections will behave under these heavier loads in terms of it's fatigue life. In order to best meet this need this research paper has considered not only current standard loads but also future potential loads.

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Some of the heaviest locomotives in the North American railway industry can reach up to 435,000 pounds and generally consist of 2 ~ three axle sets spaced roughly 45.62 to 54.63' apart (see Figure 26) [2]. Locomotives are typically found in sets of two at the front and rear of a train. Many foreign railroads already employ much larger and heavier locomotives in their fleets than do American and Canadian railroads and it is not unreasonable to assume that the North American railway industry will continue to see increases in the gross weight of their locomotives in the future [3]. The stresses induced by the locomotive are one of the primary contributing factors in the accumulation of damage to the critical connections in the emergency span.



Figure 26 - Typical 435K Railway Locomotive

Prior to the introduction of 263 kip rail cars in the 1960s the primary concern for fatigue damage to railway bridges came from the locomotive only. With the introduction of the heavier 263 kip gross weight rail cars the number of cycles accumulated in some bridge members increased by a factor of 60 [1] (* Part 9.1.3.13 commentary). The resulting damage to railway structures was severe. In the late 80s and early 90s the Association of American Railroads conducted a research study known as the Heavy Axle Load Research Program (HAL) into to the viability of using 286 kip gross vehicle weight

cars on existing railway infrastructure. The study concluded that the added strain on the railway infrastructure, and particularly bridges, and the associated maintenance cost and shorter lifespans were offset by the economic benefits of carrying heavier rail cars including those up to 315,000 pounds. This gave the industry incentive to increase loads despite the increased damage that would accumulate on railway bridges [4]. While the industry has yet to fully adopt the 315 kip rail cars as a standard for the majority of traffic the conversion to 286 kip cars is undeniable. 90% of newly acquired rail cars are rated for at least 286 kip loads and almost 100% of coal traffic and 40% of general freight traffic is moved in 286 kip or higher cars [3]. One segment of rail traffic where 315K railcars are being quickly adopted is on the so called "Coal Corridors". Therefore, this research assumed that the conversion to the heavier 315K railcars will continue and used the weight and configuration of one of the most common types of 315K cars in the FEA analysis. Figure 27 illustrates the geometry of a typical 315 kip rail car.



Figure 27 - Typical 315K Rail Car

For the purposes the FEM analysis the loads and load patterns of the 435 kip locomotive and the 286 or 315 kip railcar were broken down into individual vertical loads acting on each rail from each wheel. Together these elements make up the live load that was applied to the FEM Model. Figure 28 shows an example of the force distribution conventions used for the model. This figure illustrates the load pattern for 1~435K locomotive and 2~315K railcars. Note that even though the locomotive is heavier than the railcars it exhibits less force than the railcar per axle/wheel. A cross section of the span is shown in Figure 29 that illustrates the loading from one single axle of a 315K railcar on the temporary structure. Utilizing the moving load capability of RISA 3D the live load as well as all other moving loads were built into the program and stepped in increments of 1 inch and solved statically to determine stresses and forces at multiple points. Although at first glance a static solution might seem to be inadequate for the purposes of modelling stresses due to train loads it has been shown through extensive research [1] [5] that the fatigue behavior of steel railway bridges can be accurately modelled when corrected for dynamic behavior with appropriate static loads that will be discussed in the following sections of this research. Figure 30 shows how live loads were entered into the FEM model in order to quickly calculate the stresses and forces in the structure at multiple points.



Figure 28 - Force Distribution for Locomotive and 2 Railcars



Figure 29 - Live Load from 315K Railcar





Live Load Entry

In practice as well as in the initial phase of the FEM model it is considered that each train consists of 2 ~ 435K locomotives at the front, 110 railcars of either 286K or 315K gross weight and 2 additional locomotives in the rear. Justification for these figures is taken from the American Association of Railroads Factbook [6]. It is important to note that the Cooper E loading that is prescribed in the AREMA manual for the design of steel railway bridges and was used to design the emergency span has been shown to be inaccurate at predicting the fatigue behavior of steel structures, particularly short spans [1] [2]. This fact can be seen clearly from examination of Figure 31 and Figure 32. It is obvious that the Cooper loads would be insufficient at predicting the fatigue behavior of the span under load from railcars because the constant distributed load used to model railcars would essentially produce no effective stress range (ΔS_{re}). This discrepancy is explained by the fact that when the Copper's E loading was accepted as a standard for the railway industry in the early-20th century the use of railcars with sufficient weight to induce fatigue damage to structures had yet to see wide adoption [1] (Part 9 commentary). Furthermore it can be observed on short spans that the Cooper load underestimates the number of fatigue cycles due to the locomotive significantly and that for longer spans it is over conservative in terms of stress magnitude [2].



Figure 31 - E80 vs Actual Load - Flexure

25' Span [2]



60' Span [2]

It is for this reason that this research has concentrated on developing a fatigue analysis based on real world loads.

3.1.3 Impact Load

The amplification of loads due to impact is perhaps one of the most complicated and least understood aspects of design and analysis of railway structures. The two major impact factors that were considered during the analysis of the temporary span were vertical effects due to superstructure-vehicle interaction and the amplification of vertical wheel loads due to lateral rocking effects.

The determination of the impact factor due to rocking effect is fairly straight forward. Locomotives and railcars will tend to rock or sway due to track conditions, wind, super-elevation transition and stiffness variance in suspension systems causing vertical amplification of the wheel loads. Numerous test conducted by AREA in 1949 and 1955 have demonstrated that the rocking effect is independent of the velocity of the train [9]. AREMA has recommended that rocking effect be calculated by the following simple equation [1];

$_{RE} - \frac{100}{100}$	(Equation 5)
S	

Where,

RE = Rocking effect (as a percentage of live load)

S = Distance between centerline of rails (feet)

For most main line track on class 1 railways in North America the distance between rails can be considered approximately 5 feet meaning that the rocking effect will generally be taken as 20% of live load.

The vertical effects of superstructure-vehicle interaction are much more complex to determine. Superstructure vibration is induced by the train as it traverses the bridge. For bridges with relatively short spans in comparison to the train geometry such as the temporary span it is possible to consider the effects of concentrated moving loads to approximate an impact factor [1]. A dynamic impact factor (I_{vs}) can be determined classically as a ratio of dynamic deflection (y_d) to static deflection (y_s). Assuming small deformation and considering internal damping negligible in comparison to the external damping the partial differential equation of motion can be expressed as [2];

$$p(x,t) = EI \frac{\partial^4 y_d(x,t)}{\partial x^4} + m \frac{\partial^2 y_d(x,t)}{\partial t^2} + c \frac{\partial y_d(x,t)}{\partial t}$$
(Equation 6)

Where,

 $y_d(x,t) = Dynamic deflection at distance x and time t$

EI = Flexural stiffness of the structure

m = mass of the superstructure per unit length

c = viscous damping coefficient of the superstructure

p(x,t) = Moving load on bridge at distance x and time t

Assuming that the temporary span is sufficiently short for a dynamic response to be approximated by a concentrated load;

$$p(x,t) = \delta(\xi)P$$
 (Equation 7)

Where,

 $\delta(\xi)$ = The Dirac Delta function and physically represents the density of the idealized point mass at ξ = x-Vt

P = Concentrated force.

Considering the d'Alembert's principle of inertial effects the concentrated load can be taken as [2];

$$P = (F(t) + m_v g) - m_v [g - \left(\frac{d^2 y_d(Vt, t)}{dt^2}\right)]$$
(Equation 8)

Where,

F(t) = The dynamic force due to concentrated moving load such as vehicle suspension

 m_v = The mass of the concentrated force

g = The acceleration due to gravity

Combining (Equation 6), (Equation 7) and (Equation 8) yields.

$$\delta(x - Vt)(F(t) + m_v g) - m_v \left[g - \left(\frac{d^2 y_d(Vt, t)}{dt^2} \right) \right]$$

$$= EI \frac{\partial^4 y_d(x, t)}{\partial x^4} + m \frac{\partial^2 y_d(x, t)}{\partial t^2} + c \frac{\partial y_d(x, t)}{\partial t}$$
(Equation 9)

In order to simplify (Equation 9) the dynamic load due to suspension will be neglected due to the relatively slow 45 mph speed restriction on the temporary span. Despite this simplification the solution to the above equation is still quite complex. A complete explanation of the solution for (Equation 9) can be found in *Dynamics of Railway Bridges* by Ladislav Fryba [10]. Using the transformative techniques described by Dr. Fryba yields the following solution for dynamic displacement.

$$y_{d}(x,t) = \frac{2m_{v}gL^{3}}{\pi^{4}EI} \sum_{i=1}^{\infty} \frac{1}{i^{2}[i^{2}\left(i^{2}-\left(\frac{\omega}{\omega_{1}}\right)^{2}\right)^{2}+4\left(\frac{\omega}{\omega_{1}}\right)^{2}\left(\frac{\omega}{\omega_{1}}\right)^{2}]} \begin{bmatrix} i^{2} \begin{bmatrix} i^{2} & 10 \end{bmatrix} \\ -\left(\frac{\omega}{\omega_{1}}\right)^{2} \end{bmatrix} sini\omega t \\ -\frac{i\left(\frac{\omega}{\omega_{1}}\right)\left[i^{2}\left(i^{2}-\left(\frac{\omega}{\omega_{1}}\right)^{2}\right)-2\left(\frac{\omega}{\omega_{1}}\right)^{2}\right]}{\sqrt{i^{4}-\left(\frac{\omega}{\omega_{1}}\right)^{2}}}e^{-\omega t}sin\sqrt{\omega_{i}^{2}-\omega_{c}^{2}t} \\ -2i\left(\frac{\omega}{\omega_{1}}\right)\left(\frac{\omega}{\omega_{c}}\right)\left(cosi\omega t-e^{-\omega_{c}t}xcos\sqrt{\omega_{i}^{2}-\omega_{c}^{2}t}\right) \end{bmatrix} sin\frac{i\pi x}{L}$$

Where,

$$\omega_1 = \frac{\pi^2}{L^2} \sqrt{\frac{EI}{m}}$$
 is the fundamental frequency of an unloaded simply supported beam [2]
[10]

$$\omega = \frac{\pi V}{L}$$
 is the forcing frequency of p(x,t)

In the case of the temporary structure it can be assumed that $\omega_c < 1$ and neglected [2]. Furthermore, previous equations can be simplified by neglecting the damping and setting c=2m ω_c =0. Assuming a sinusoidal shape function as prescribed by Anil Chopra in *Dynamics of Structures* [11] yields the following equation for mid-span deflection.

$$y_d(midspan, t) = \frac{2P}{mL\left[\omega_1^2 - \left(\frac{\pi V}{L}\right)^2\right]} (\sin\frac{\pi Vt}{L} - \frac{\pi V}{\omega_1 L} \sin\omega_1 t)$$
(Equation 11)

Figure 33 shows the calculated dynamic deflection at mid-span of the 55 foot tangent substructure emergency span configuration using (Equation 11) vs the static deflection at mid-span.



Figure 33 - Dynamic Deflection of Temporary Span

The calculated maximum deflection due to dynamic interaction of a 315K railcar and superstructure interaction on inspection of Figure 33 is 1.039". The deflection due to static forces is also shown in Figure 33 and is calculated as follows;

(Equation 12)

$$y_s = \frac{PL^3}{48EI} = .908"$$

As stated previously the impact factor due to vehicle-superstructure interaction can be classically solved as a ratio of dynamic to static deflection.

$$I_{vs} = \frac{y_d}{y_s} = \frac{1.039}{.908} = 1.145$$
 (Equation 13)

(Equation 13) indicates that the impact factor due to superstructure-vehicle interaction is roughly 14.5%. However, due to the simplifying assumptions made in the formulation of (Equation 11) the final calculated impact factor will likely underestimate the forces due to impact as experienced by the actual structure significantly. An understanding of all the characteristics that govern the magnitude of the impact factor of vehicle-superstructure interaction is important to understanding why the calculated impact could be significantly undersized. The primary parameters affecting the dynamic behavior of the emergency beam span are;

- Train Speed
- Span Length
- Track Geometry
- Mechanical condition of the train
- Live load characteristics of the train

- Train handling
- Bridge supports
- Bridge layout
- Condition of the track

AREMA recommends a formula for the determination of an appropriate impact factor based on empirical analysis [1]. On spans less than 80 feet the vertical impact factor recommended by AREMA is.

$$I_{\nu s} = 40 - \frac{3L^2}{1600}$$
(Equation 14)

For the temporary structure with a length of 55 feet, $I_{vs} = 40 - 3(55)^2/1600 =$ 34.33%. When combined with the effects due to rocking the total impact load for the temporary structure was taken as $I_{total} = 34.33+20 = 54.33\%$. Figure 34 shows how the impact factor was applied to the FEM model.



Figure 34 - FEM Impact Factor

The large discrepancy between the calculated impact factor due to 315K railcars and the empirical impact factors described in AREMA at first seem incompatible. However, it must be noted that the AREMA impact factors have a probability of occurrence of approximately 1%. Even at that low probability of occurrence it is assumed that a permanent railway structure that is designed for at least 80 years of service will be subjected to this full impact load multiple times within it service life. For the purposes of fatigue analysis the AREMA manual allows a reduction in the empirical impact factor to 35% for beams such as the beam emergency span (Figure 35) [1]. Applying this reduction to the empirical impact value yields an AREMA assumed mean impact value for the temporary steel span of 12.02%. This is much more in line with the calculated value of 14.5%.

Member	Percentage			
Members with loaded Lengths				
\leq 10 feet (3m) and no load sharing	65%			
Hangers	40%			
Other Truss members	65%			
Beams, Stringers, Girders and Floor Beams	35%			
Note: Where bridges are designed for operation of trains handling a large percentage of cars with flat or out of round wheels which increase impact and/or poor track which increases impact, and the loaded length of the member is less than 80 feet (24m), the mean impact should be 100% of the design impact.				

Figure 35 - Fatigue Impact Load Percentages [1]

However, the reductions assumed in Figure 35 are not appropriate for the temporary span due to the multiple unknown variables associate with its implementation such as and the condition and type of track it will be used on as well as the variability of
the span length and geometry. Full impact load was used for the FEM analysis in this study.

3.1.4 Centrifugal Forces

Centrifugal forces are the product of train loads as they travel through a curve in the track and act through the centroid of the train's mass in a direction perpendicular to the direction being travelled. The AREMA design manual recommends that the center of gravity be taken at a distance 8 feet above the rail situated at an equal distance between the rails (see Figure 37) [1]. A quick inspection of the geometries of the 435K locomotive and the 315K rail car shown in Figure 26 and Figure 27 will confirm that this is a valid recommendation for the geometries being considered for this research study. The centrifugal force corresponding to each axle load is shown below [2].

$$CF_A = \frac{P_A * V^2}{g * R}$$
(Equation 15)

Where,

 $P_A = Axle load (kips)$

V = Speed of the train (mph)

- g = Acceleration due to gravity = 32.174 ft/s/s = 78,972 mph/hr
- R = radius of the curve = 5730/D (ft) = 1.085227/D (mi)

Converting all variables into units of kips, miles and hours and plugging into (Equation 15) yields;

$$CF_A = \frac{P_A * V^2 * D}{78,972 * 1.085227}$$
(Equation 16)
= 0.0000117P_A V^2 D

Where,

D = Degree of Curvature

Now consider that during the design of the temporary span it was decided that the maximum allowable curvature (D) would be limited to 2 degrees and that the maximum allowable speed as outlined in the section devoted to impact loads would be set to 60 mph. Plugging these values into (Equation 16) and considering the axle load of a typical 315K railcar will give an example of an appropriate centrifugal load as applied at the center of gravity (see Figure 37).

 $CF_A = 0.0000117 * 78.75 * 60^2 * 2 = 6.63 kips$

In order to apply the centrifugal load to the FEM model in a way that would simulate the load acting through the center of gravity it was necessary to perform some simple statics calculations to come up with a series of loads that would have an equivalent effect on the structure. Figure 36 shows the static solution for the reactions at the rails for the centrifugal force from 1 axle of a typical 315K railcar. The equivalent loading at the rails (which are the point of load transfer to the structure from the railcar) to simulate the centrifugal force is equal and opposite of the reactions calculated in the static solution. Figure 39 shows how these forces are applied to the temporary span.



$$\sum M_B = 0$$

(6.63)(8)+R_{AY}(-5)=0
=> R_{AY}=10.61 Kips

$$\sum F_{Y} = 0$$

10.61+R_{BY}=0
=> R_{BY}=-10.61 Kips

$$\sum F_X = 0$$

R_{AX}+ R_{AX} -6.63=0, R_{AX}= R_{BX}

 $=> R_{AX}$ and $R_{BX}=3.315$ Kips

Figure 36 - Free Body Diagram of Centrifugal Force







Showing Centrifugal Forces Entry



of Gravity



Figure 39 - Equivalent Centrifugal Force System on FEM Model

3.1.5 Longitudinal Forces

The dynamic forces applied to the emergency beam span in the longitudinal direction are primarily a product of braking and tractive forces [1] [2]. There have been a number of changes to the recommendations for the magnitude and placement of the longitudinal forces within the AREMA manual over the last several decades. By the time of the 1996 edition of the design manual it was shown through testing conducted by the AAR that longitudinal forces were in fact as much as 25 times greater than those recommended in the manual [1] [7]. A precise analysis of the longitudinal effect of train loads is very complex and continues to be an active field of research within the railway engineering community. For the purposes of this research the most current recommendations in the AREMA manual were used and adapted to suit needs of the FEM analysis.

The 2013 edition of the AREMA MRE was developed for use with the Cooper's E80 load and states that the longitudinal force shall act at 3' above the rail and be taken as;

 $LF = 25\sqrt{L}$

(Equation 17)

Where,

L = Length of Bridge

For use with other loading conventions the manual allows for (Equation 17) to be scaled appropriately. An example of how this load was modelled for 315K railcars on a 55' span configuration is shown below.

 W_{E80} = Total weight of E80 locomotive = 568 kips

 $W_{315} = Gross weight of 315K railcar = 315 kips$

$$LF = \frac{W_{315}}{W_{E80}} * 25\sqrt{L}$$
 (Equation 18)

For a 55' span configuration;

$$LF_{55s} = \frac{315}{568} * 25\sqrt{55} = 102.8 \text{ kips per car}$$

Dividing this number by the number of axles per car (4) and the number of wheels per axle (2) shows that the longitudinal force per wheel on a 55' skew span is equal to 12.85 kips.

Chapter 4

FEM Results from Macro Analysis Using RISA 3D

An iterative analysis of the multiple combinations of loads and geometries applicable to the determination of the effective lifespan of the temporary structure was undertaken in order to assess which combination proved to be the controlling set. From early on in the analysis it became clear that the longest span length that the structure was designed for (55') was going to produce the most damage due to fatigue. The primary factor that makes shorter spans more fatigue prone is the increased number of load cycles that shorter spans experience relative to much longer spans. However, when considering the locomotive and railcar geometries outlined in previous sections and running through a series of moving loads in the FEM models it was discovered that even at 55' the temporary span experienced one load cycle per railcar. This fact coupled with the additional stresses induced at the critical connection caused by a longer span geometry it was decided to focus the remainder of this research on the 55' condition.

Extensive modelling was done of the temporary span with a variety of substructure alignments in order to study the effects of a skewed substructure on the stresses at the critical connection (see Figure 40). It was determined by inspection of the results of the analysis that the effects on the induced stresses were largely negligible and that the maximum deflection differential for the skewed span was actually somewhat less

than that of the tangent substructure span. As a consequence the 55' tangent substructure span was used as the bases for the ensuing analysis.



Figure 40 - 55' Span - Skew Substructure

The final condition considered was the effect of the loading geometry and magnitude on the stresses at the critical connection. As discussed previously the 286K railcar and the 435K locomotive are industry standard on North American railway mainlines with the 315K railcar transition currently underway. It is intuitively obvious that the 315K car would produce greater stresses in the critical member than the 286K railcar but what proved somewhat surprising is that the 315K railcar proved to have a comparable effect on the structure as the 435K locomotive. This is because the 315K car produces a heavier load per axle than the locomotive and the length of the span limits the number of axles that will be acting on the structure at a time. Considering that the railcars will induce over 100 more cycles for a typical train than will the locomotives it was decided for the purposes of this research to consider the 315K railcar as the loading condition for the subsequent steps in the analysis.





Figure 41 - FEM Model 55' – Tangent

The moving load capabilities of RISA 3D were utilized in order to determine appropriate forces at the critical section of the temporary structure. The output data from this analysis was interpolated into a cycle history of the induced forces in order to apply these conditions to a more detailed Abaqus CAE model of the welded connection. Figure 42 illustrates the output data that was generated by RISA 3D and Figure 43 shows the position of loading at the time of maximum and minimum moment.

🗼 Envelope Member Section Forces															
	Member	Sec		Axial[k]	LC	y Shear[k]	LC	z Shear[k]	LC	Torque[lb-ft]	LC	y-y Moment[lb-ft]	LC	z-z Moment(Ib-ft)	LC
1289		5	max	-2.724	1-2601	23.999	1-1170	9.609	1-1147	18676.146	1-1160	53587.334	1-1145	8898.853	1-2601 .
1290			min	-11.186	1-761	-54.004	1-847	1.215	1-66	-19135.416	1-748	1175.269	1-1	-1.15448e+6	1-944
1291	M130	1	max	-2.576	1-2601	21.162	1-1184	9.363	1-1147	18392.693	1-1160	63193.643	1-1147	8671.321	1-2601
1292			min	-10.593	1-761	-59.527	1-862	1.154	1-66	-19507.901	1-776	-356.959	1-1	-1.15539e+6	1-944
1293		2	max	-2.576	1-2601	21.162	1-1184	9.363	1-1147	18392.693	1-1160	64363.981	1-1147	8765.06	1-2601
1294			min	-10.593	1-761	-59.527	1-862	1.154	1-66	-19507.901	1-776	-206.543	1-1	-1.15179e+6	1-945
1295		3	max	-2.576	1-2601	21.162	1-1184	9.363	1-1147	18392.693	1-1160	65534.32	1-1147	8858.799	1-2601
1296			min	-10.593	1-761	-59.527	1-862	1.154	1-66	-19507.901	1-776	-56.127	1-1	-1.14835e+6	1-946
1297		4	max	-2.576	1-2601	21.162	1-1184	9.363	1-1147	18392.693	1-1160	66704.659	1-1147	8952.538	1-2601
1298			min	-10.593	1-761	-59.527	1-862	1.154	1-66	-19507.901	1-776	94.289	1-1	-1.14509e+6	1-947
1299		5	max	-2.576	1-2601	21.162	1-1184	9.363	1-1147	18392.693	1-1160	67874.997	1-1147	9046.277	1-2601
1300			min	-10.593	1-761	-59.527	1-862	1.154	1-66	-19507.901	1-776	244.705	1-1	-1.14213e+6	1-949
1301	M131	1	max	-2.602	1-2601	33.401	1-1181	7.013	1-1213	983.753	1-2485	247.149	1-1	9046.234	1-2601
1302			min	-10.615	1-761	-47.325	1-859	1.111	1-66	-46147.845	1-891	-36252.487	1-1147	-1.14213e+6	1-949
1303		2	max	-2.602	1-2601	33.401	1-1181	7.013	1-1213	983.753	1-2485	468.791	1-1	9203.786	1-2601
1304			min	-10.615	1-761	-47.325	1-859	1.111	1-66	-46147.845	1-891	-34957.277	1-1147	-1.1413e+6	1-952
1305		3	max	-2.602	1-2601	33.401	1-1181	7.013	1-1213	983.753	1-2485	690.432	1-1	9361.338	1-2601
1306			min	-10.615	1-761	-47.325	1-859	1.111	1-66	-46147.845	1-891	-33662.068	1-1147	-1.14118e+6	1-955
1307		4	max	-2.602	1-2601	33.401	1-1181	7.013	1-1213	983.753	1-2485	1023.899	1-67	9518.89	1-2601
1308			min	-10.615	1-761	-47.325	1-859	1.111	1-66	-46147.845	1-891	-32366.858	1-1147	-1.14153e+6	1-957
1309		5	max	-2.602	1-2601	33.401	1-1181	7.013	1-1213	983.753	1-2485	1453.711	1-67	9676.442	1-2601
1310			min	-10.615	1-761	-47.325	1-859	1.111	1-66	-46147.845	1-891	-31071.648	1-1147	-1.14216e+6	1-958
1311	M132	1	max	-2.454	1-2601	29.113	1-2378	6.767	1-1213	794.784	1-2485	-314.152	1-1	9448.912	1-2601
1312			min	-10.023	1-761	-50.45	1-874	1.05	1-66	-46525.782	1-891	-22243.977	1-1147	-1.14307e+6	1-958
1313		2	max	-2.454	1-2601	29.113	1-2378	6.767	1-1213	794.784	1-2485	36.02	1-1	9665.207	1-2601
1314			min	-10.023	1-761	-50.45	1-874	1.05	1-66	-46525.782	1-891	-20162.216	1-1147	-1.13742e+6	1-960
1315		3	max	-2.454	1-2601	29.113	1-2378	6.767	1-1213	794.784	1-2485	554.767	1-67	9881.503	1-2601
1316			min	-10.023	1-761	-50.45	1-874	1.05	1-66	-46525.782	1-891	-18080.454	1-1147	-1.1327e+6	1-964
1317		4	max	-2.454	1-2601	29.113	1-2378	6.767	1-1213	794.784	1-2485	1232.661	1-67	10097.799	1-2601
1318			min	-10.023	1-761	-50.45	1-874	1.05	1-66	-46525.782	1-891	-15998.693	1-1147	-1.12981e+6	1-968
1319		5	max	-2.454	1-2601	29.113	1-2378	6.767	1-1213	794.784	1-2485	2152.069	1-898	10314.094	1-2601 .

Figure 42 – Results of RISA 3D Analysis



Figure 43 - Position of Loading at Max and Min Moment Along Primary Axis

The load cycle prediction of the forces applied to the critical connection was achieved by interpolating data from the RISA analysis and assuming a sinusoidal shape for the loading history by applying variables to (Equation 19).

$$y(t) = Asin(2\pi ft + \varphi) + D$$

(Equation 19)

Where

A = The amplitude or peak deviation of the function from zero.

f = The ordinary frequency

t = Time in seconds

 φ = Phase in radians

D = A non-zero center amplitude

The ordinary frequency was computed by assuming the train is traversing the span at the maximum allowable speed of 45 MPH and relating this velocity to the static analysis steps in the RISA model. Table 4 shows an example of the tabulated load history for the W36x262 beam at the critical connection.

	W36x262 Beam								
Time (Sec)	Moment M _{z-z} (lb-ft)	Abaqus Amplitude Factor (Mz- z)	Moment M _y . (lb-ft)	Abaqus Amplitude Factor (My-y)	Shear V _y (lb)	Abaqus Amplitude Factor (Vy)			
0.0000	-236,279.54	0.207	50285.77	0.741	-21162.00	0.356			
0.0014	-232,024.93	0.203	49990.95	0.737	-21160.35	0.355			
0.0028	-227,797.75	0.199	49694.55	0.732	-21155.39	0.355			
0.0043	-223,598.33	0.196	49396.62	0.728	-21147.12	0.355			
0.0057	-219,427.03	0.192	49097.19	0.723	-21135.54	0.355			
0.0071	-215,284.18	0.188	48796.28	0.719	-21120.67	0.355			

Table 4 - Load History for W36x262 Beam

Table 4 - continued

0.0085	-211,170.13	0.185	48493.92	0.714	-21102.48	0.355
0.0099	-207,085.21	0.181	48190.14	0.710	-21081.00	0.354
0.0114	-203,029.76	0.178	47884.97	0.705	-21056.21	0.354
0.0128	-199,004.10	0.174	47578.45	0.701	-21028.13	0.353
0.0142	-195,008.58	0.171	47270.60	0.696	-20996.75	0.353
0.0156	-191,043.51	0.167	46961.45	0.692	-20962.07	0.352
0.0170	-187,109.23	0.164	46651.03	0.687	-20924.11	0.352
0.0185	-183,206.04	0.160	46339.38	0.683	-20882.86	0.351
0.0199	-179,334.29	0.157	46026.52	0.678	-20838.32	0.350
0.0213	-175,494.27	0.154	45712.49	0.673	-20790.50	0.349
0.0227	-171,686.31	0.150	45397.31	0.669	-20739.40	0.348
0.0241	-167,910.72	0.147	45081.01	0.664	-20685.04	0.347
0.0256	-164,167.81	0.144	44763.64	0.660	-20627.40	0.347
0.0270	-160,457.88	0.140	44445.21	0.655	-20566.50	0.345
0.0284	-156,781.25	0.137	44125.77	0.650	-20502.34	0.344
0.0298	-153,138.20	0.134	43805.33	0.645	-20434.93	0.343
0.0312	-149,529.04	0.131	43483.94	0.641	-20364.27	0.342
0.0327	-145,954.07	0.128	43161.62	0.636	-20290.37	0.341

Given the fact that the critical connection will be subject to complex multi-axial loading it was important to ensure that all loads would be sequenced appropriately within the FEM analysis on the detailed connection in Abaqus. To facilitate this it was necessary to develop a special amplitude factor to use in the Load and Step modules in Abaqus as shown in Table 4 that would not only approximate a sinusoidal load pattern but also ensure that all loads are applied to the model in the correct magnitude relative to time (*t*). Figure 44 shows the coordinate system conventions that were used in the RISA 3D analysis of the temporary span.

Figure 45 represents the moments at the critical connection of the temporary beam span plotted as a function of time in seconds. Figure 46 shows the shear in the web of the rolled I beam at the location of the critical connection as a function of time in seconds.







Figure 45 - Moments in 55' Beam (W36x262)



Figure 46 - Shear in 55' Beam (W36x262)

In addition to the forces acting on the beam along its primary axis there will also be significant stresses on the critical connection induced in the out of plane direction due to differential deflection of the beams. The centrifugal and wind forces explained in previous sections will induce an uneven load distribution on each of the W36x262 beams that could lead to relatively significant differences in their maximum defections at the location of the critical connection. Figure 47 shows how this deflection causes moment forces and out of plane stresses in the temporary span.



Figure 47 - Differential Deflection of W36x262

	Diaphragm					
Time Seconds	Differential Deflection (Δ)	Abaqus Amplification Factor (Δ)				
0.0000	-0.00447528	0.213				
0.0014	-0.00439767	0.209				
0.0028	-0.00432055	0.206				
0.0043	-0.00424395	0.202				
0.0057	-0.00416785	0.198				
0.0071	-0.00409228	0.195				

Table 5 – Differential Deflections

The out of plane forces induced by differential deflection are transferred to the beam through the diaphragm and stiffener. An inspection of Figure 47 shows that a tensile stress concentration is likely to be highest at the lowest point of the welded stiffener to web connection in Beam 2. This will increase the maximum principle stresses and reduce the fatigue life of the connection in a way that is not directly referenced in the AREMA design manual. In order to model the complex geometry and

account for this phenomenon a more detailed FEM model will be used in the following sections.

As with the other forces acting on the critical connection a sinusoidal deflection history of the differential deflection at the critical connection was assumed and an Abaqus Amplification Factor was calculated in order to facilitate modelling in Abaqus (see Table 5). The deflection histories were plotted as a function of time and are shown in Figure 48.



Figure 48 - Differential Deflection over Time

The finite element analysis capabilities of RISA 3D proved very useful as a means of quickly modelling and obtaining useful results on the response of the emergency span to the complex loading conditions required to accurately approximate real world load cases. The moving load feature proved especially well-suited to analyzing the structure in a useful way on a macro level. However, in order to obtain a more detailed understanding of the local response of the critical connection to the loads it will be subjected to it was necessary to constructed a more detailed finite element model and apply the loads obtained from the macro analysis.

Chapter 5

Modelling of Critical Connection Using Abaqus CAE

Using the values calculated from the macro analysis a three dimensional representation of the critical connection was modelled using Abaqus CAE. The combination of forces and displacements obtained from the methods outlined in the above sections was applied to this model in order to ascertain the stress data that was needed for a comprehensive fatigue analysis. Two stress states are required for a comprehensive fatigue life of the critical connection; ΔS_{re} – the stress range at the critical location due to applied loads and σ_{res} – the residual stress at the critical location due to applied loads and σ_{res} – the residual stress at the critical location and assumed defect were then used to determine appropriate stress intensity factors utilizing a finite element methodology for the applied loads and a weight function for the residual stresses. This information was in turn used in a linear elastic fracture mechanics analysis outlined later in this research in order to determine an effective fatigue life.

5.1 Stresses Due to Applied Loads

The critical connection was modelled as a thin strip of the W36x262 beam with the stiffener and diaphragm modelled in their entirety. The beam was subjected to the forces calculated from the RISA 3D analysis applied to the cut face of the beam. The moment applied to the section was modelled as a linearly variable distributive load

utilizing a user defined formula to achieve appropriate distribution of stresses. The Abaqus amplitude factors outlined in Table 4 were used in the definition of the amplitude of the load over time in order to assure a sinusoidal load history on the connection. The diaphragm was fixed at one end and a deflection in the Y direction was induced in the beam using a similar methodology (amplitude factors for sinusoidal history) as the loads in order to model the effects of differential deflection over time. The loads were applied in a step module in Abaqus with the incrimination parameters set up to emulate the time for one complete cycle on the structure.

Figure 49 shows a contour plot of the Mises stress history of the connection over a period of time. The region of concern in terms of the fatigue life of the critical connection is the area at the base of the weld. It is this area that experiences the greatest magnitude of tensile stress along the length of the weld. When considering only in plane forces it is obvious that the extreme fiber of the beam will experience a greater magnitude of tensile stress than the bottom of the weld connection but it is assumed that there is much less likelihood of a defect capable of causing crack propagation being present in the un-welded portion of the member. This coupled with the fact that out of plane stresses are introduced by the stiffener to web connection made the area at the base of the weld a primary concern.



Figure 49 - Stress Distribution at Critical Connection

The connection was modelled using 3D stress quadratic tetrahedral elements (C3D10 shown in Figure 50) with a refined mesh at the base of the weld (see Figure 49).



Figure 50 - C3D10 Element



Figure 51 - Principle Stresses near the Weld Toe over Time

Figure 51 shows the principle stresses near the weld toes over time and gives a good indication of the effect out of plane stresses have on the magnitude of the stress range. The measured stress range at the extreme fiber of the welded connection in the FEM analysis is more than triple the allowable stress range for the fatigue category C' in

the AREMA manual and is also more than double the anticipated stress range obtained using the Cooper E loading with a strictly in plane analysis. However, inspection of the data showed that the stress range evens out fairly quickly to more moderate levels the further away from the extreme bottom of the weld toe the stresses are examined. Since the region with the highest stress range is also the region most likely to develop fatigue cracks these results alone can validate the need for this research as a tool for future use of the emergency span.

5.2 Residual Stresses Induced by Weld

It has been extensively proven as shown in the literature [15] [16] that residual stresses introduced by arc welding processes have a direct effect on the fatigue life of a structural member. The use of finite element simulations to predict residual stresses in as welded members is the subject of active study and has largely gained the acceptance of the engineering community as a reasonably accurate prediction tool [18]. A sequentially coupled finite element analysis was undertaken on the fillet weld of the stiffener to web connection of the critical connection in order to ascertain the appropriate stresses (and stress intensity factors using a weight function as outlined in later sections) associated with the process. The following assumptions were made as to the type and process of the weld procedure;

- Shield Metal Arc Welding (SMAW)
- 5/16" Weld bead Single Pass
- Direct Current

• All field welded connections

The SMAW process is the most commonly used type of field weld in the railway industry and is assumed to be the welding process of choice for the type of rapid construction needs associated with the emergency span erection. In the SMAW process an electric current from a welding power supply is used to form a current between the base material and a consumable electrode. The electric current, or arc, between the base material and the electrode creates the extremely high temperatures (up to 6500°F at the tip) needed to fuse the connection. The electrode is coated in flux which forms protective gasses and slag in order to protect the weld zone from atmospheric contamination. The high heat induced by the weld arc causes the base material to deform in a plastic manner in the heat affected zone by introducing large stresses and strains as well as causing thermal expansion. After the weld cools the member doesn't completely return to its original shape causing distortion induced stress fields. Residual stress fields in tension can have a significant negative effect on the fatigue life of welded connections while compressive residual stress can have a positive impact [19]. In order to facilitate the efficient modelling and to keep model complexity reasonable a sub-section of the critical connection was chosen for modelling as a separate analysis. Figure 52 shows the section of model that was considered for this welding analysis. In order to best understand the overall FEM processes and procedures an overview of heat transfer and residual stress fundamentals as they relate to this research analysis are briefly covered in the next two sections.



Figure 52 - Section of Connection Modelled for Weld Analysis

5.2.1 Thermal Heat Effects Caused by Welding

Heat flow as passed from the arc or torch to the base material is of principle concern in the study of a thermal FEM analysis. In arc welding the total heat output is a product of the current and the voltage when considering a direct current. This can be expressed as [15];

$$Q = VI\eta_{heat}$$

(Equation 20)

Where

V = Voltage

I = Current

 η_{heat} = Process dependent heat efficiency

For the Purposes of FEM modelling the heat generated can be expressed as:

 $Q = Q_S + Q_V + Q_f \tag{Equation 21}$

Where

 $Q_S = Surface heat$

 $Q_V = Volume heat$

 Q_f = Filler heat generated by the addition of the bead material

For the FEM analysis to output a usable heat history the heat fields are needed as a function of time and space. In order to describe the temperature fields in an isotropic continuum element with temperature dependent material properties over time the following equation is utilized in the FEM formulation;

$$\frac{\partial T}{\partial t} = \alpha \left(\frac{\partial^2 T}{\partial x^2} + \frac{\partial^2 T}{\partial y^2} + \frac{\partial^2 T}{\partial z^2} \right) + \frac{1}{c\rho} \frac{\partial Q_V}{\partial t}$$
(Equation 22)

Where,

 $\frac{\partial T}{\partial t}$ = is the rate of change in temperature at a point over time

T = T(x,y,z,t) is temperature as a function of space and time

 $\rho = \text{density}$

 $\alpha = \frac{k}{c\rho}$ is the thermal diffusivity

K = Thermal conductivity

C = specific heat capacity

When a usable heat transfer analysis over time is compiled with the finite element method the results are written to and input file that can be used in a subsequent stress analysis.

5.2.2 Residual Stresses

Residual stresses are primarily caused by the variation of temperature across a region or volume whose magnitude can be calculated using the above equations. These stresses, termed thermal stresses, will dissipate as the temperature variance evens out. In the case of welded steel connections the heat variance becomes high enough for plastic deformation to occur. Due to thermal expansion in the heated region an elastic compressive zone is formed with the surrounding colder material. When the plastically compressed region cools it gives way to a tension stress field. It is these stress fields that will have a direct negative effect on the fatigue life of the temporary span.

5.2.3 FEM Procedure

The FEM analysis as performed using Abaqus software was approached using the sequentially coupled Thermal-Stress analysis method discussed above. Weld beads were modelled by partitioning the three dimensional region of the part meant to represent the weld material. The weld arc was simulated by applying a user defined temperature of 2740°F (which is the melting temperature of A572 Gr 50 Carbon Steel) at the boundary between the partitioned beads and the interaction surfaces of the model. The beads were activated in the model sequentially over time by applying an Abaqus "model change" in a separate step for that bead. The beads only interact with beads from previous steps that have been activated in an Abaqus model change but beads that have yet to be activated are still allowed to deform with the model in order to avoid convergence issues. These beads had no effect on the stiffness of the system since they were considered to be near their melting point and therefore very ductile. The arc/torch application was analyzed through transient heat transfer steps and then a cool down step was introduced that deactivated the boundary conditions associated with the heating step. The values for the heat transfer were then fed into the static stress analysis as a loading history. The coupled analysis relies heavily on temperature dependent material properties and complex boundary conditions that can be researched extensively by an examination of the literature [20] [21] [22] [23]. Figure 53 shows a schematic of the modelling process and analysis steps.



Figure 53 - Schematic of FEM Modelling Process for

Welded Connection

5.2.4 FEM Results

The results from the FEM analysis of the web to stringer welding process were examined and compared to a range of studies that presented empirical data on welded connection residual stresses as well as some FEM based analysis to ensure that a reasonable result was obtained [14] [15] [18] [25] [26] [27]. One of the driving goals of

the FEM analysis was to obtain the transverse residual stresses induced at the weld toe of the connection as these stresses will play a significant role in the estimation of fatigue life as discussed in later sections of this paper. Results were plotted both with regard to time and distance (in inches) away from the weld toe using the conventions outlined in Figure

54.





Distance Results





Figure 56 - Temperature as a function of time for sample elements along the X-axis

Results for longitudinal (Figure 58) and transverse (Figure 60) residual stresses show good agreement with neutron defraction measurements and finite element method predictions as presented in the literature for similar type weldments [14] [27]. The transverse residual effects are considered to control for the purposes of fatigue estimation for the geometry of the critical connection as defects are most likely to coincide with the edge of the weld toe and propagate normal to the transverse stress field. Figure 56 shows the temperature in the connection at several elements over time. The analysis considered that one side of the stiffener would be welded before the other and, as a result, even though the analysis considers a single weld pass the area around the weld would be heated twice as the arc moved across the opposite side. The results in the time period 400 – 700 in Figure 56 clearly demonstrate the effect this assumption has on the analysis.



Figure 57 - Misses Stresses at Weld



Figure 58 - Longitudinal Residual Stress along X-axis

The behavior of the stress distribution showed a good agreement with the principles of residual stress behavior as outlined earlier in this section. Figure 58 shows the tension zone near the weld toe giving way to a compressive residual stress field further along the x-axis. The transverse residual stress plot over time shown in Figure 59 also indicates good agreement with the anticipated behavior of the weld by showing that peak compressive stresses as the weld arc passes over the element give way to a reversal into a tensile stress field as the cooling step takes effect.

The size of the sub-model and the boundary conditions that were used in the FEM analysis were carefully chosen in order to balance accurate results with the complexity and time constraints associated with running the analysis. The stiffness of the system has a direct effect on the concentration of residual stress as thermal expansion generates compressive stresses in the surrounding material. If a system is too stiff it will show unrealistically high residual stresses. The cross section of the model was carefully chosen to ensure that these forces had enough volume to dissipate heat energy at an acceptable level, especially in the transverse direction (see Figure 55). The model is likely stiffer in the longitudinal direction than real world conditions which may lead to somewhat higher longitudinal stresses than would be experienced in the field. However, the predicted results still fall within a reasonably accurate range of expected results.



Figure 59 – Transverse Residual Stress near Weld Toe over Time



Figure 60 - Transverse Residual Stress along X-axis

Chapter 6

Fatigue Life Assessment Utilizing LEFM

Linear elastic fracture mechanics (LEFM) is the study of crack propagation within solids subjected to stresses that produce a linear elastic response. It is an extremely important tool for determining the fatigue life of details with pre-existing defects. In order to gain a better understanding of the basic concepts associated with the implementation of LEFM and how it was utilized in the context of this research study a general overview of LEFM is presented.

6.1 Overview of Linear Elastic Fracture Mechanics

LEFM is an analytical approach to crack propagation that relates the stress field magnitude and distribution near a crack tip to the applied stresses, size / shape / orientation of the crack, and material properties. The fundamental principle of LEFM is that the stress field ahead of a crack tip can be characterized by a stress intensity factor (SIF). In order to analyze the stresses at the crack tip in an elastic solid it is common practice in classic LEFM to define the relative movements of two crack surfaces as modes. Theses modes represent the local displacements in an infinitely small element containing a crack singularity and are shown in Figure 61.



Figure 61 - Modes of Crack Surface Displacement

Mode I is characterized by a tensile induced opening of the crack surfaces relative to one another and is usually the driving mode in terms of crack behavior for most engineering applications [19]. Mode II is the result of in plane shearing of the surfaces and Mode III is characterized by out-of-plane shearing effects. The stress intensity factor for each of the three failure modes can be expressed as a limit with r approaching zero and stress as a function of r and θ as shown in (Equation 23), (Equation 24) and (Equation 25).

$$K_{I} = \lim_{r \to 0} \sqrt{2\pi r} \ \sigma_{yy}(r,\theta)$$
(Equation 23)

$$K_{II} = \lim_{r \to 0} \sqrt{2\pi r} \ \sigma_{yx}(r,\theta)$$
(Equation 24)
$$K_{III} = \lim_{r \to 0} \sqrt{2\pi r} \ \sigma_{yz}(r,\theta)$$

Where r and θ are shown in Figure 62 relative to the coordinate system and stress components;



Figure 62 - Stress Components and Global Coordinates ahead of a Crack

The range in stress intensity factor at a defect ($\Delta K_{I, II, III}$) is one of the driving parameters affecting the propagation of cracks and can be expressed by (Equation 26).

$$\Delta K_{I,II,III} = K_{I,II,III max} - K_{I,II,III min}$$

(Equation 26)

(Equation 25)

Where,

 $K_{I,II,III max}$ = The maximum stress intensity factor under cyclic loading for modes I, II, and III

 $K_{I,II,III min}$ = The minimum stress intensity factor under cyclic loading for modes I, II, and III

A method for relating the change in SIF to crack propagation rate and the number of cycles to failure was developed by P. C. Paris in the early 1960s and is the most widely used method for determining fatigue life using LEFM [28]. The equation proposed by Paris is known as the *Paris-Ergodan Power Law* and can be expressed as;

$$\frac{da}{dN} = C\Delta K^m \tag{Equation 27}$$

Where,

a = Crack length

N = Total number of constant amplitude stress range cycles

m = Material constant derived from regression analysis and taken as m = 3 for structural steel

C = Coefficient of crack growth

Integration of (Equation 27) yields;

(Equation 28)

$$N = \frac{1}{C} \int_{a_i}^{a_f} \frac{da}{\Delta K^m}$$

Where,

 a_i = Initial crack length

 a_f = Final crack length

Crack growth can be divided into three growth regions based on stress intensity parameters (see Figure 63). It is assumed that there is a region where the change in SIF (ΔK) will not be of sufficient magnitude to propagate a crack within the context of LEFM. This region is referred to as Region I. Region II is the region that is of concern when utilizing LEFM and constitutes the region in which cracks are assumed to propagate at a constant rate when subjected to constant amplitude cyclic loading. The value separating these two regions is known as the threshold value (ΔK_{th}). Any applied SIF range below this threshold will not cause appreciable crack growth behavior. Region III deals with rapid fracture of the crack material and occurs when ΔK exceeds the fracture toughness range (ΔK_c) of the material.

This study covers only the behavior of the material in Region II with Region I stresses considered to have infinite fatigue life and Region III stresses considered to have experienced structural failure.



Figure 63 - Crack Growth Regions

Where there exists a significant interaction between fracture modes I, II, and III the crack growth can be characterized as mixed mode. All of the methods put forward that are known to this author to account for the effects of mixed mode fatigue cracking are based on the Paris-Ergodan Law (Equation 27). The interaction of the fracture modes is still a field of active study and progress but for the purposes of this research the equation for an equivalent SIF range (ΔK_{eq}) as proposed by Tanaka [24] was used.

$$\Delta K_{eq} = \left[\Delta K_I^4 + 8 * \Delta K_{II}^4 + \frac{8 * \Delta K_{III}^4}{(1-\nu)} \right]^{0.25}$$
(Equation 29)

The equivalent factor (ΔK_{eq}) can then be substituted into (Equation 27) in order to calculate the cycles (N) to failure for the critical span under applied load.

6.2 Defects in the Critical Connection

Defects that arise as a consequence of the welding procedure have a significant effect on the fatigue life of any welded structure. Some of the most common defects associated with welded connections are cold laps, weld to cracks and root defects. Of these defects it has been shown in various literature [12] [13] that as many as 80% of initial welding defects are cold laps. Cold laps form when the toe of a weld remains cold enough that it doesn't fuse to the base metal and the weld simply overlaps the base material. This leaves no continuity between the two surfaces and allows for a concentration of stresses to build up at the tip of the overlap (see Figure 64). Cracks at the toe or root of the weld are another common type of defect and are mainly caused by transverse shrinkage due to improper pre-heating, an influx of hydrogen in the weld pool by way of moisture in the HAZ and a sufficient amount of transverse residual stress. For this research it is assumed that some form of weld defect i.e. crack is present at the critical connection prior to the emergency span being released into service. For the purposes of fatigue life calculation this assumption should prove to be conservative. However, it should be noted that the majority of the fatigue life in as welded connections is consumed by the propagation phase [14]. The assumption of an initial crack being present at the weld toe is particularly appropriate when considering that the critical connection is intended to be welded in the field in an emergency situation. The weld site will not be in a controlled environment and will be open to the elements, thus increasing the possibility of moisture related cracking issues. The skill level and experience of the welder coupled with a less strenuous inspection procedure are other variables that will

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affect the quality of the weld in terms of possible defect. This research has assumed an initial defect in the weld toe with a length of approximately 0.05 inches (1 mm) and a depth of 0.03 inches (0.76 mm). These estimates are consistent with studies of defect measurements in the literature and should prove to be conservative in terms of size and placement [12] [13].



Figure 64 - Weld Defects

In order to determine the fatigue life in the critical connection the stress intensity factors at the tip of the assumed crack were calculated for both the applied and residual stresses as outlined in the next sections.

6.3 Stress Intensity Factor at the Critical Connection Due to Applied Loads

In order to calculate the SIF due to applied load at the critical connection a global model was used to drive the stresses in a sub-model with an assumed defect of dimensions 0.05 inches long and 0.03 inches in depth (see Figure 65). The crack was modelled near the toe of the weld with the weld to base material transition given a 0.01 inch fillet to help avoid singularity issues in the stress formulation. The results of the analysis show excellent agreement with the predicted behavior of stress and strain fields

corresponding to the shape and size predicted in the literature [19] [14] [15]. The FEM formulation for the SIFs was carried out in FRANC3D utilizing the Interaction Integral and Displacement Correlation Methods, both of which showed good agreement with one another.



Figure 65 - Refined Mesh at Point of Crack

Figure 66 - Plastic Zone at Tip of Crack

(Surface)

Figure 66 shows the plastic zone that forms at the tip of the crack at the surface of the model and Figure 67 shows the plastic zone at the crack tip for the through material thickness defect. The crack was modelled as a semi-elliptical toe/root crack (see Figure 68) cutting perpendicular to the surface of the web as opposed to a cold lap that would have had an initial alignment parallel to the surface. The root crack is expected to have a more rapid fatigue growth rate and as such provided for more conservative results. In order to predict the crack growth rate on the surface of the web as well as in the through plate direction it is necessary to solve the SIFs for the entire length of the defect. Figure 69 and

Table 6 show the values for the SIF along the length of the crack.





Figure 68 - Cross Section of Defect





Figure 69 - Plot of Stress Intensity Factors for the Assumed Defect

N Coordinates	KI	KII	KIII	Х	Y	Z
0.009	12251.11	475.2239	-543.044	-2.26	-15.2001	-0.4189
0.0268	11947.41	194.3362	-720.966	-2.26	-15.2004	-0.4166
0.0444	11713.5	-24.8045	-853.479	-2.26	-15.2009	-0.4144
0.0618	11561.72	-249.667	-949.634	-2.26	-15.2017	-0.4124
0.0793	11465	-470.987	-1010.42	-2.26	-15.2027	-0.4104
0.0968	11396.07	-681.811	-1044.32	-2.26	-15.2039	-0.4085
0.1144	11345.44	-881.45	-1060.52	-2.26	-15.2052	-0.4067
0.1321	11325.94	-1062.91	-1055.91	-2.26	-15.2067	-0.405
0.1498	11318.07	-1224.16	-1035.07	-2.26	-15.2084	-0.4034
0.1676	11323.64	-1368.23	-1004.39	-2.26	-15.2101	-0.402
0.1852	11335.6	-1503.35	-965.084	-2.26	-15.2119	-0.4006
0.2026	11360.21	-1626.31	-920.413	-2.26	-15.2137	-0.3994
0.2203	11412.59	-1733.34	-875.54	-2.26	-15.2157	-0.3982
0.2384	11457.95	-1831.33	-836.26	-2.26	-15.2177	-0.3971
0.2565	11483.79	-1921.5	-791.977	-2.26	-15.2198	-0.3961
0.2746	11514.29	-1998.91	-744.236	-2.26	-15.2219	-0.3952
0.2926	11545.24	-2063.92	-692.608	-2.26	-15.224	-0.3944
0.3107	11573.7	-2124.88	-639.592	-2.26	-15.2262	-0.3936
0.3287	11589.05	-2178.88	-587.442	-2.26	-15.2284	-0.3929
0.3468	11603.12	-2236.66	-529.857	-2.26	-15.2306	-0.3923
0.3649	11623.23	-2290.25	-470.243	-2.26	-15.2329	-0.3918
0.3829	11633.07	-2335.55	-412.252	-2.26	-15.2351	-0.3914
0.401	11654	-2372.62	-356.274	-2.26	-15.2374	-0.391
0.4191	11662.35	-2407.54	-301.246	-2.26	-15.2397	-0.3907
0.4371	11673.23	-2431.59	-244.212	-2.26	-15.242	-0.3904
0.4551	11674.88	-2448.87	-187.038	-2.26	-15.2443	-0.3902
0.4731	11670.39	-2466.05	-132.452	-2.26	-15.2466	-0.3901
0.491	11662.46	-2475.65	-78.6377	-2.26	-15.2489	-0.39
0.509	11661.6	-2477.65	-10.0873	-2.26	-15.2511	-0.39
0.527	11678.03	-2477.82	40.2342	-2.26	-15.2534	-0.3901
0.5451	11679.34	-2465.76	93.4779	-2.26	-15.2557	-0.3902
0.5631	11662.22	-2450.88	146.3889	-2.26	-15.258	-0.3904
0.5812	11640.82	-2423.32	199.5429	-2.26	-15.2603	-0.3907
0.5993	11621.3	-2392.22	258.3933	-2.26	-15.2626	-0.391
0.6174	11608.25	-2359.35	318.4164	-2.26	-15.2649	-0.3914
0.6355	11592.64	-2324.71	378.7105	-2.26	-15.2672	-0.3918
0.6535	11563.81	-2287.34	441.38	-2.26	-15.2694	-0.3924
0.6716	11552.97	-2237.72	503.6691	-2.26	-15.2716	-0.393
0.6896	11530.17	-2180.07	563.1044	-2.26	-15.2738	-0.3936
0.7076	11500.83	-2116.56	619.931	-2.26	-15.276	-0.3944
0.7257	11483.68	-2048.95	675.0282	-2.26	-15.2781	-0.3952
0.7437	11466.77	-1970.61	728.47	-2.26	-15.2802	-0.3961
0.7617	11448.76	-1881.56	775.9277	-2.26	-15.2823	-0.3971
0.7798	11412.44	-1775.97	814.4756	-2.26	-15.2843	-0.3982
0.7975	11361.73	-1660.31	860.7024	-2.26	-15.2863	-0.3994
0.8149	11330.91	-1526.59	908.3927	-2.26	-15.2881	-0.4006
0.8326	11325.48	-1388.5	947.3525	-2.26	-15.2899	-0.402
0.8504	11327.23	-1238.87	977.2256	-2.26	-15.2917	-0.4034
0.8681	11326.85	-1076.42	995.8143	-2.26	-15.2933	-0.405
0.8858	11334.3	-896.945	1002.612	-2.26	-15.2948	-0.4067
0.9035	11368.93	-705.81	989.4372	-2.26	-15.2961	-0.4085
0.921	11436.21	-503.286	957.9431	-2.26	-15.2973	-0.4104
0.9384	11533.29	-291.945	900.0378	-2.26	-15.2983	-0.4124
0.9558	11680.56	-/3.6223	807.0489	-2.26	-15.2991	-0.4144
0.9733	11910.48	138.7492	679.6187	-2.26	-15.2996	-0.4166
0.991	12213.54	405.697	507.1156	-2.26	-15.2999	-0.4189

Table 6 - Stress Intensity Factors and Coordinates

6.4 Stress Intensity Factor at the Critical Connection Due to Residual Stresses

The SIF due to residual stresses (K_{res}) was calculated using a weight function and incorporated into the LEFM fatigue analysis by way of the mean stress method. This approach makes the determination that transverse residual stresses acting normal to the assumed defect will be the predominate form of residual stress driving crack growth. (Equation 30) was used to calculate the stress intensity factor due to residual stress (K_{res})

$$K_{res} = \sigma_{res} \sqrt{\frac{\pi a}{Q}} * h(a, t)$$
(Equation 30)

Where,

- σ_{res} = Residual stress affecting crack propagation and taken as
- a = Depth of assumed crack into base material
- t = Thickness of the steel web
- h(a, t) = Weight function
- $Q = \Phi_0^2$ With the value taken from Barson and Rolfe [19] Table 2.13 page 47

For a more detailed explanation of the weight function assumed for this analysis as well as a list of weight functions used for various geometries please examine the work of Barsom and Rolfe [19] and Fett et al [29] [30] [31]. The stress intensity factor due to residual stresses was then incorporated into the LEFM analysis by way of the mean stress method using the following equation for R greater than or equal to 0.

$$\frac{da}{dN} = \frac{C(\Delta K)^3}{\sqrt{1-R}}$$
(Equation 31)

Where,

R = The load ratio. Taken as $\frac{K_{eq min} + K_{res}}{K_{eq max} + K_{res}}$

6.5 Fracture Toughness, Coefficient of Crack Growth and Critical Crack Length

Before the Paris Ergodan Miner Law could be utilized to solve for fatigue life at the critical connection the Paris Law parameters (C and m) and final crack length (a_f) were required. The final crack length (a_f) was considered to occur at a point where the crack transitioned from stable to unstable growth or in other words moved from region II to region III growth characteristics (see Figure 63). The length of crack required for this to occur is known as the critical crack length and is a function of the fracture toughness and the design stress present in the connection [19]. For the purposes of this research it is considered that;

$$a_f = a_c$$
 (Equation 32)

Where,

 a_c = The critical crack length

The critical crack length for the critical connection can be found using the following equation [19].

$$a_c = \frac{1}{\pi} \left(\frac{K_{IC}}{\sigma_{max}} \right)^2$$
(Equation 33)

Where,

 σ_{max} = The maximum stress present at the critical location

 K_{IC} = The critical fracture toughness of the material

Note that (Equation 33) is valid for a crack that propagates along the face of the steel surface. The crack assumed at the critical connection will almost certainly spread in this manner; however, the crack will likely propagate into the thickness of the material as well. For the purposes of this research this author has elected to consider the thickness of the W36x262 web (0.84") to be the critical crack length for a defect that propagates normal to the plate surface. This approach is consistent with the best practices for fatigue control within the railway industry [1] [2].

The value of K_{IC} is typically determined by a testing method that accounts for the toughness directly. However it is possible in many cases to estimate the value of K_{IC} using energy absorption data from a test such as the Charpy V-notch test (CVN) when no other means are available [33]. The Chary V-notch test is a standardized high strain-rate

test that yields information on a material's notch toughness by measuring the amount of energy absorbed by a material during brittle fracture. It is usually given with results for three separate impacts. Table 7 shows the Charpy V-notch data obtained for the W36x262 A572 Gr. 50 steel beams used for the emergency span.

Table 7 - Charpy V-notch Data for Emergency Span

	Mechanical Properties						Chemical Properties															
Yield Tensile	Yield Strength	Tensile Strength	ELONG	Temp	Imap	Imapact Energy		C Ma		D		ci	C 11	Cu Ni	Cr	Mo	v	Cb	CE	Sn	Pcm	СІ
Ratio	KSI	KSI	%	F	FT-LBS				P	3	21	Cu										
	Мра	Мра	%	С	JOULES																	
0.92	62	76	28.1	10 F	86	90	94	0.07	1 25	0.016	0.010	0.26	0.22	0.27	0.4	0.04	0.04	0.002	0.01	0.41	0 10	6 5
0.82	427	524	28	-12 C	117	127	121	0.07	1.25	0.016	0.018	0.20	0.55	0.27	0.4	0.04	0.04	0.002	0.01	0.41	0.19	0.5

Rolfe and Barsom et al proposed that the following equation can be used to estimate the value of K_{IC} using Chary energy data [34].

$$\left(\frac{K_{IC}}{f_y}\right)^2 = \frac{5}{f_y} \left(CVN - \frac{f_y}{20}\right)$$
(Equation 34)

Where,

 f_y = The yield strength of the steel material

CVN = Charpy V-notch energy absorption

Conservatively using a value of $f_y = 50$ ksi (62 ksi would also be acceptable

based on Table 7) and utilizing (Equation 34) gives a value of for K_{IC} of 144.5 ksi \sqrt{in} .

Plugging the value for K_{IC} and the stress values for the critical connection shown in Figure 51 into (Equation 33) yields a value for a_c and a_f of 3.77 inches which is consistent with experimental data found in Barsom and Rolfe [19].

Another variable that needed to be addressed before a fatigue analysis became possible was the establishment of a quantifiable connection between steel microstructure and crack growth. It has been shown in the literature [19] [34] that the fatigue crack growth behavior in steels can be subdivided into three main microstructural groups; ferrite-pearlite, martensitic and austentic. The crack growth behavior in each of these steel microstructure groups can be estimated to a reasonable degree of accuracy regardless of any variation in tensile strength and chemical composition [19]. The A572 Gr. 50 structural steel considered for the construction of the emergency span falls into the ferrite-pearlite category along with most structural steels.

Like all steels, ferrite-pearlite steels are created by adding a small amount of carbon to iron in order to produce a material with superior mechanical properties. Since the carbon atom's atomic diameter is less than the interstices of the iron crystal lattice the carbon will dissolve into the iron lattice and cause a distortion of the crystalline structure of the iron atoms when heated during the fabrication process (see Figure 70). Ferrite-pearlite steel is then produced by allowing the heated steel to cool at a very slow rate during the manufacturing process. This allows ferrite-pearlite steel to exhibit a much finer microstructure than steels that are cooled more rapidly. It is this microstructural density that will have a direct effect on crack propagation as it pertains to this research.

The coefficient of crack growth (C) as proposed in (Equation 27) is meant to account for the effects this microstructural material property will have on crack propagation. Extensive research and testing has shown that the behavior of ferrite-pearlite steels like A572 Gr. 50 is a constant that can be taken as [19];

$C = 3.6 * 10^{-10}$



Figure 70 - Ferrite-Pearlite Steel Microstructre

6.6 Results and Number of Cycles to Failure

Using the information that was obtained from the investigation outlined in the preceding sections of this research it was possible to make a comprehensive fatigue life prediction for the temporary span utilizing (Equation 27 through 30). The crack propagation characteristics of the connection were assessed for two different crack

growth directions; along the surface of the web parallel to the weld pass and into the material thickness. The fatigue life with and without the consideration of residual stresses is presented and compared in order to effectively gage the effect that these stresses have on structural performance. Note that compression stresses induced by the cyclic loading were considered negligible in magnitude compared to the tension stresses that were developed and where taken as equal to zero (see Figure 51). This has the effect of ensuring that $\Delta K = K_{max}$ for all cases.

6.6.1 Crack Propagation along the Surface of the Web

(Equation 33) represents the crack length that would need to develop before the defect begins to exhibit region III growth behavior. As discussed in previous sections, region III behavior is considered to be structural failure in the context of this analysis. Table 8 shows the results of the fatigue analysis for the surface crack and Figure 71 shows the relationship between crack length and cycles to failure.

	SIF Ksi ($(in)^{1/2}$			Number of	Percentage			
	ΔK_{I}	ΔK_{II}	ΔK_{III}	K _{res}	Cycles (N)	Difference			
With Residual	12.251	0.475	0.543	1.762	5,254,437	6.72%			
Stress									
Without	12.251	0.475	0.543		5,619,602	6.72%			
Residual Stress									

Table 8 - Fatigue Life Results Surface for Surface Cracks



Figure 71 - Relationship between Crack Length and Cycles to Failure (Surface Crack)

6.6.2 Crack Propagation into Thickness of Material

Stress intensity factors taken from

Table **6** and located at the center of the half elliptical defect being analyzed were used to determine the propagation rate of the crack through the thickness of the steel web. The results of the analysis are listed in Table 9 and the relationship between crack length and number of cycles to failure is shown in Figure 72.

	SIF Ks	i (in) ^{1/2}	Number of	Percentage			
	ΔK_{I}	ΔK_{II}	ΔK_{III}	K _{res}	Cycles (N)	Difference	
With Residual	11.7	-2.448	-0.187	1.762	1,263,156	6.99%	
Stress							
Without	11.7	-2.448	-0.187		1,354,599	6.99%	
Residual Stress							

Table 9 - Fatigue Life Results for Through Thickness Crack Growth

Crack Propagation into Base Material



Figure 72 - Relationship between Crack Length and Cycles to Failure (Through Crack)

6.7 Discussion of Results

A number of interesting points were learned from the results obtained in this research analysis. Some of the most important takeaways from the above results are categorized below;

- Through thickness propagation of defects present at the weld toe in the form of cold laps or toe/root cracks are most likely to be the controlling design life parameter of the temporary span.
- The emergency span's expected design life is a conservative approximation of N = 1.263e6 cycles. This is less than the 2 million cycles required for a permanent railway structure according to AREMA [1].
- The effects of residual stresses on the effective fatigue life of the temporary span are relatively minimal. Examination of Table 8 and Table 9 show that the average reduction in design life when considering residual stresses are around 7% for the temporary span.

For the purposes of fatigue calculations, AREMA Committee 15 estimated an upper limit of rail traffic on any given rail line to be about 20 trains a day with 2 locomotives carrying 150 railcars [1]. This kind of heavy traffic utilizing 315K railcars is not a reality on the present day railway system but it will make for a good approximating of the absolute maximum amount of cycles that the emergency span can be feasibly subjected to in the near future. When converting those trains to equivalent cycles on the temporary span one can deduce from this research that the span can be expected to perform with confidence for a minimum of 421 days. This is more than enough time for a replacement span to be designed, fabricated and built.

It is also worth noting that the bridge engineer confronted with the challenge of having to use the emergency span will have access to a complete loading history of the structure that is to be replaced. Comparing those documents to the findings put forth in this research paper will allow the engineer to make decisions about how long the span can be left in place. In fact, on some parts of the railway system that carry lighter traffic it might be feasible to use the temporary structure as a semi-permanent replacement for the damaged bridge. This would allow for an excellent opportunity to study the response of the emergency structure in real world conditions as a means of comparison with the FEM findings.

Any use of the temporary structure on a class 1 railway mainline for a period of longer than 421 days would likely require a specialized inspection procedure. The weldments in the tension zone at the critical connection would be of primary importance in regard to inspections.

As mentioned earlier the residual stresses induced by the welding process had a surprisingly small effect on the fatigue life of the temporary span. While this is at first somewhat surprising, an inspection of Figure 58 and Figure 60 should make it clear that the orientation of the assumed crack was the primary driver in this result. The magnitude of the predicted longitudinal residual stress is much greater than the predicted transverse stress. Since mode I parameters are the preeminent driver of fatigue crack growth in the context of LEFM and the assumed crack was oriented for mode I interaction with the transverse and not longitudinal residual stresses it stands to reason that the residual stress effects would be lower. If the crack were oriented normal to the weld pass then the residual stresses would have a much larger role in

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the fatigue life while the tensile stress from the applied load would play a much smaller role.

Chapter 7

Summary, Conclusions and Future Work

7.1 Summary

The problem of providing a flexible solution to the challenge of service outages on railway lines due to vehicle or other impacts to railway bridge infrastructure is explored in this study with a particular interest on the fatigue service life of welded connections for the proposed solution. A systematic overview of the current state of temporary and rapid bridge construction techniques is undertaken and the strengths and weaknesses of these solutions are discussed. A new type of temporary span is then proposed to service the use cases that are not covered by current practices. The design, configuration and construction methods of this proposed emergency span is then detailed and examined using the current industry standard Copper-E design loads. In order to save on construction time and to accommodate skewed substructure configurations the emergency span is designed to utilize welded connections for the diaphragms and stiffener plates. It is shown that even under Cooper-E loads this detail falls outside the allowable stress range for the AREMA fatigue category to which it belongs, thus validating the need for further analysis.

Once the need for further analysis into the service life of the critical welded connection is confirmed it is proposed to use finite element analysis along with more representative loading conditions to determine the fatigue life of the connection. In order to facilitate the most efficient means of achieving this result the analysis is broken into three phases using the methods and FEA software that makes the most sense for each phase. Phase 1 constitutes a macro analysis of the structure using RISA 3D to determine the location of the critical connection and the stresses induced at the location by the prescribed loads. Phase 2 builds on the information obtained in Phase 1 to build a detailed model of the critical connection in Abaqus CAE. Phase 3 leverages the principles of linear elastic fracture mechanics in order to predict the fatigue life to the connection and make recommendation in regard to its effective use.

In order to most accurately predict the structural response of the temporary structure in Phase 1 it was decided to use loading conditions that are representative of the actual equipment that will be used on the structure both presently or in the near future (Ex: 435K locomotives and 315K railcars). This is in contrast to the AREMA loads that are based on outdated steam locomotive geometry from a time when fatigue conditions were not well understood. Once the appropriate loads were identified the temporary structure was modelled in RISA 3D in order to take advantage of the rapid modelling and moving load capabilities of the software. A loading history was then interpolated from the results of the RISA analysis to be used in the subsequent phases of the study.

Once the load history of the critical connection was established the results were fed into a more detailed model of the connection that was modelled using Abaqus CAE for Phase 2. The Phase 2 analysis consisted of two main objectives; determine the stresses in the connection due to applied loads and determine the residual stresses induced by the welding process. Two separate models of the connection were utilized for

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this purpose using 3D stress quadratic tetrahedral elements. The stresses due to applied loads were determined by applying the combination of forces and displacements calculated in the first phase using the loading history and the calculated amplitude factors to ensure appropriate distribution of stresses. To determine the residual stresses due to shield metal arc welding a sequentially coupled finite element analysis was performed on a smaller section of the connection that was modelling using Abaqus CAE.

The final phase of this research uses linear elastic fracture mechanics to predict the growth of a crack from a preexisting defect that was conservatively assumed to have occurred during the welding process before the span was released into service. A sub model of the defect was subject to the forces due to applied loads calculated in phase 2 in order to determine the stress intensity factor. The stress intensity factor due to residual stresses was calculated using a weight function and along with the SIF obtained for the applied stresses was incorporated into the LEFM analysis by way of the mean stress method and the *Paris-Ergodan Miner Law* to predict crack propogation rates and number of cycles to failure.

The number or cycles to failure was calculated both with and without the residual stress effects in order to determine the magnitude of influence the residual stresses have on the fatigue life of the structure. Using conservative information prescribed by AREMA regarding the maximum train geometry (number of locomotives and railcars that make up a train) and frequency of train traffic on any given rail line the calculated cycles to failure were converted into a timeframe for which the temporary span can be reasonably expect to perform without fatigue failure (421 days).

7.2 Conclusions

Finite element analysis is a useful tool for predicting the behavior of structures under load without the need for costly and time consuming physical testing. The use of the FEM is gaining acceptance as a valid means to predict the real world response of structures before they are built and tested. This trend is contrary to the traditional FEM approach of testing first in the lab and then confirming the results with FEM analysis. This research has proposed to predict the fatigue life of a critical connection within an emergency span designed to be used on a class 1 railway mainline. The use of FEM modelling as well as the principles of linear elastic fracture mechanics made it possible to take into account many complex processes associated with the design and construction of the span including a prediction of the effect of residual stresses induced by arc welding of the critical connection in the tension zone.

Many of the parameters used to obtain the final results of this research were conservative to a fairly significant degree. The impact loads introduced to the FEM model to simulate the dynamic impact of the vehicle-superstructure interaction were particularly conservative by as much as 65%. Also the likelihood of having a poorly maintained track on a curve with a relatively high operation speed of 45 mph as assumed in this research is extremely unlikely. The decision to consider a pre-existing defect at the toe of the weld at the location of critical stress concentration within the member can

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also be considered as conservative. These analysis decisions were made in order to give the bridge engineer considering the use of the emergency span a greater degree of confidence in the structure as a semi-permanent solution to a bridge hit under any conceivable confluence of circumstances. For example, if a bridge hit occurs on a short line with a poorly maintained track then the bridge engineer can rely on the higher impact factor assumed in this research to justify its use. Also if the bridge emergency were to occur on a mainline track with a 2 degree curve and a 45 mph operating speed with good track conditions then the engineer can still maintain confidence that the emergency span can be used in such a situation without making any additional changes to the design of the span. It is for this reason that this research constitutes a worthwhile endeavor that if considered properly in the context of the safe implementation of an emergency span can help to facilitate increased railway traffic velocity and positively affect the railway's core business.

7.3 Future Work

Future work on this subject is needed in order to verify the results obtained in the analysis and to gain a better understanding of actual field conditions the temporary span will be subjected to. Some of the measures that can be taken in the future are as follows;

• Utilize strain gages to determine the actual impact on an open deck steel beam span over time. This can be used to either validate the assumptions used in this research or show a need for a re-examination of the results.

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- In the event of the need to implement the emergency span for a bridge hit, strain gages could be used at the critical connection to compare the structures actual stress range with that predicted using FEM.
- Utilize electrochemical fatigue sensor technology to study the propagation of cracks in members with similar details. If the emergency span is brought into service for any significant period of time then inspection of the connection for weld defects and the utilization of EFS sensors to study crack propagation would be useful in determining the accuracy of the FEM analysis.
- Study the microstructural changes that can occur in ferrite-pearlite steel as a result of the welding process and investigate the effect that these changes can have on the crack growth coefficient and propagation characteristics.

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Biographical Information

Curtis Heinsen received his Bachelor of Science from the University of Texas at Arlington in 2009 and a Master of Science in Civil Engineering with a specialty in structures and applied mechanics from the University of Texas at Arlington in 2015.

He currently works in the field of railway bridge engineering and his research interests include steel fatigue analysis in structures, weld analysis using FEM, and dynamic impact analysis on railway bridges.